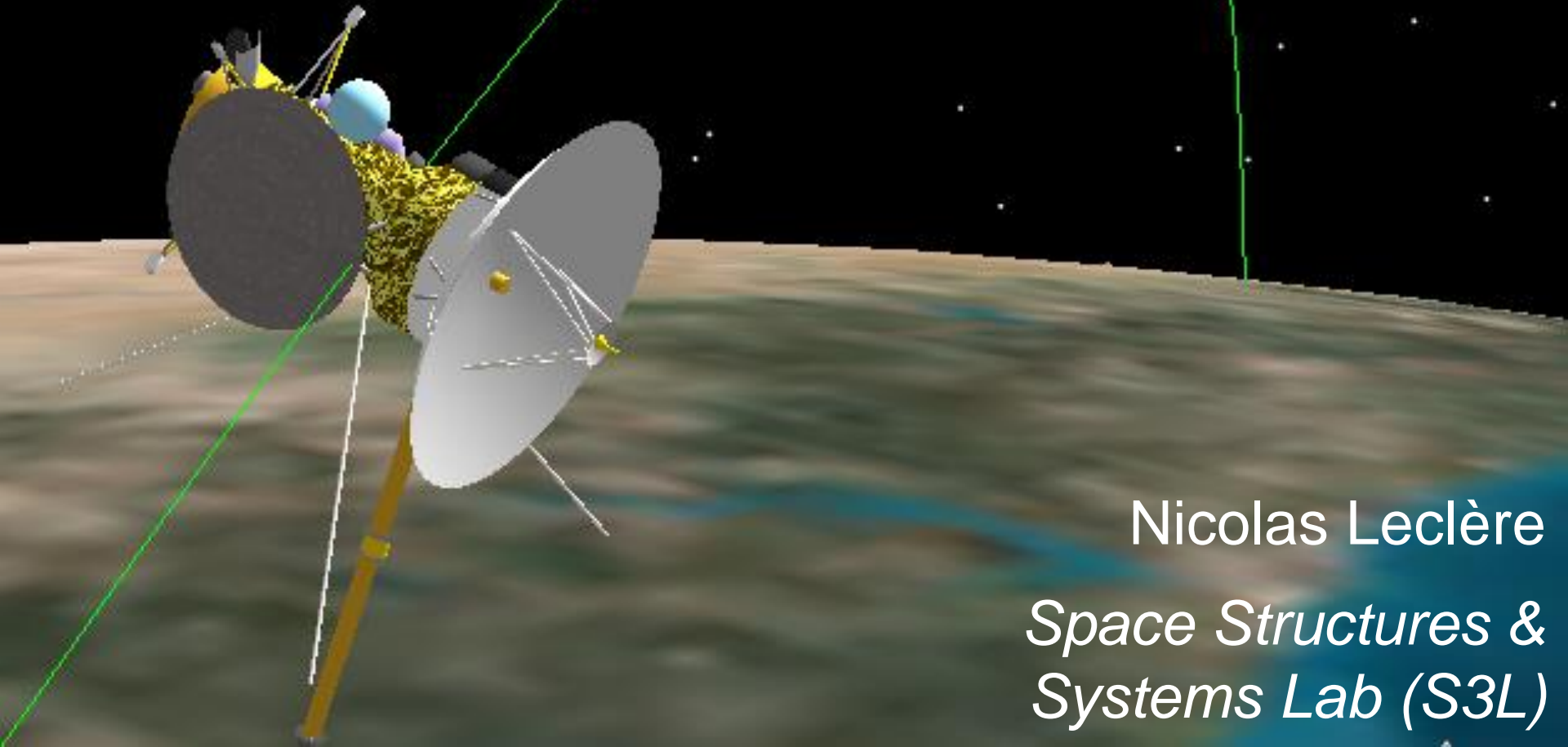


Cassini Classical Orbit Elements
Time (UTCG): 15 Oct 1997 09:18:54.000
Semi-major Axis (km): 6685.637000
Eccentricity: 0.020566
Inclination (deg): 30.000
RAAN (deg): 150.546
Arg of Perigee (deg): 230.000
True Anomaly (deg): 136.530
Mean Anomaly (deg): 134.891

Aerodynamics

(AERO0024)

9. *Small Bodies Exploration*



Nicolas Leclère
*Space Structures &
Systems Lab (S3L)*

Small bodies in the Solar system

Asteroids



Asteroids are small rocky celestial objects. They are much smaller than planets, between 100m and 1000km.

Most of them are situated in the main asteroid belt between Mars and Jupiter.

There are more than 1,000,000 asteroids in the Solar system.

Small bodies in the Solar system

Comets



Frozen bodies composed of dust, rock and ices.

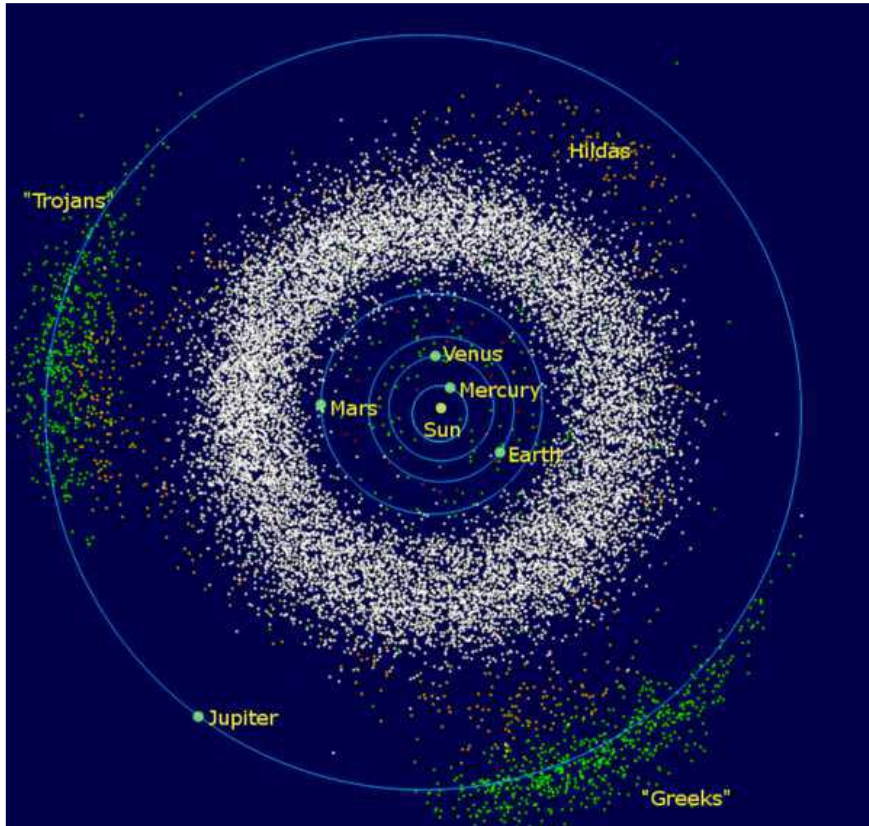
When they orbit close to the Sun, they heat up and form a tail made of gases and dust.

Comets dimension varies between 100m and 30km.

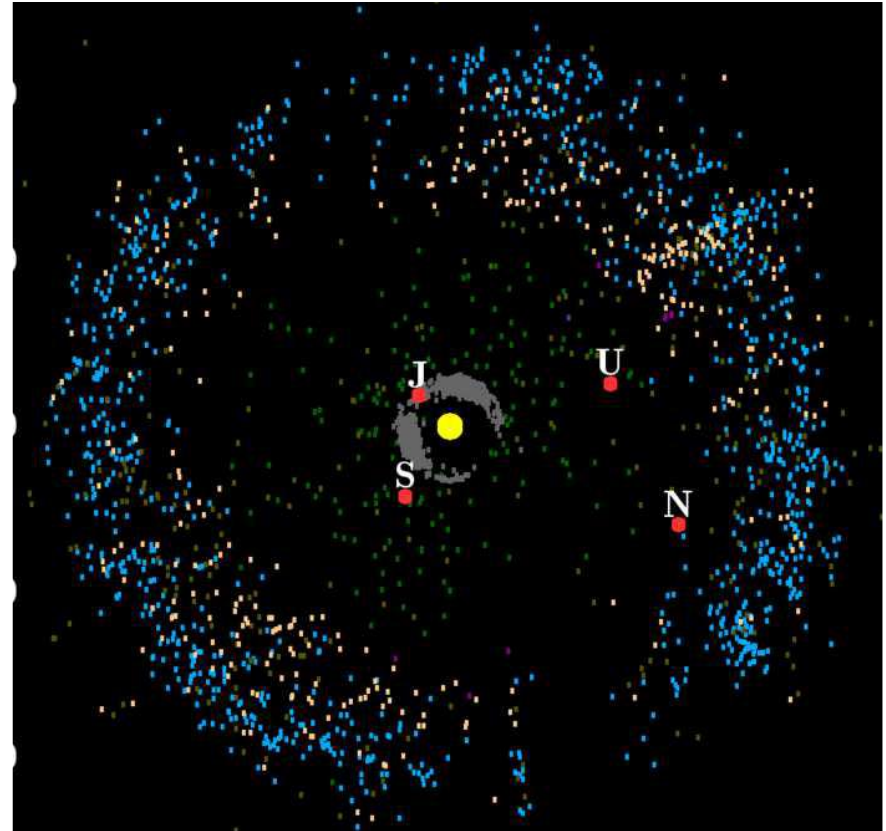
The current number of known is over 3500

Small bodies in the Solar system

Inner solar system

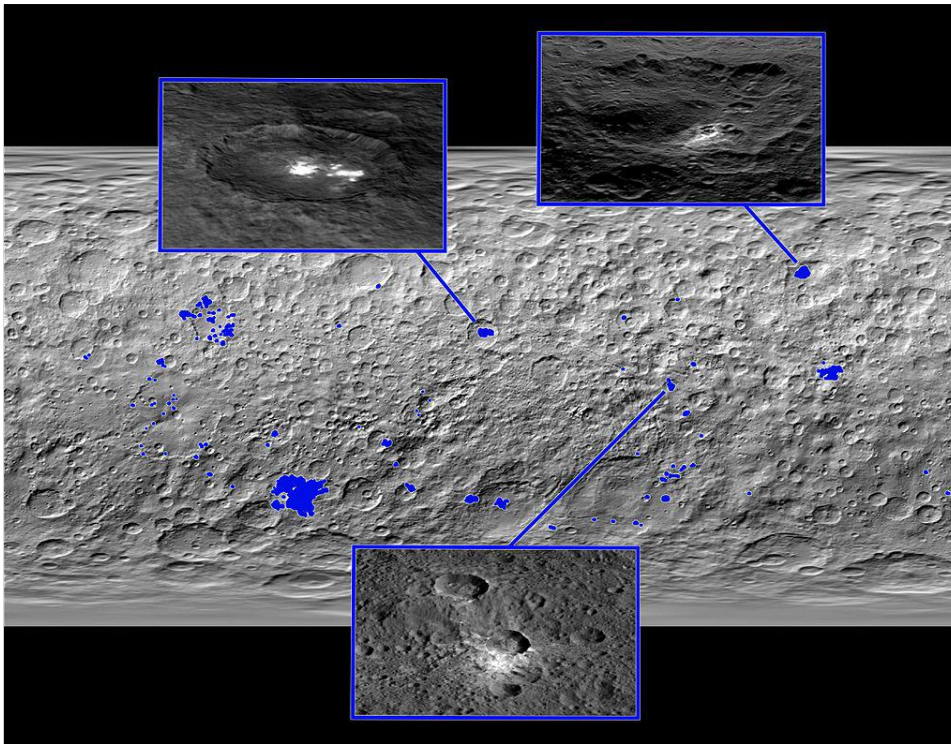


Outer solar system



Small bodies exploration, why?

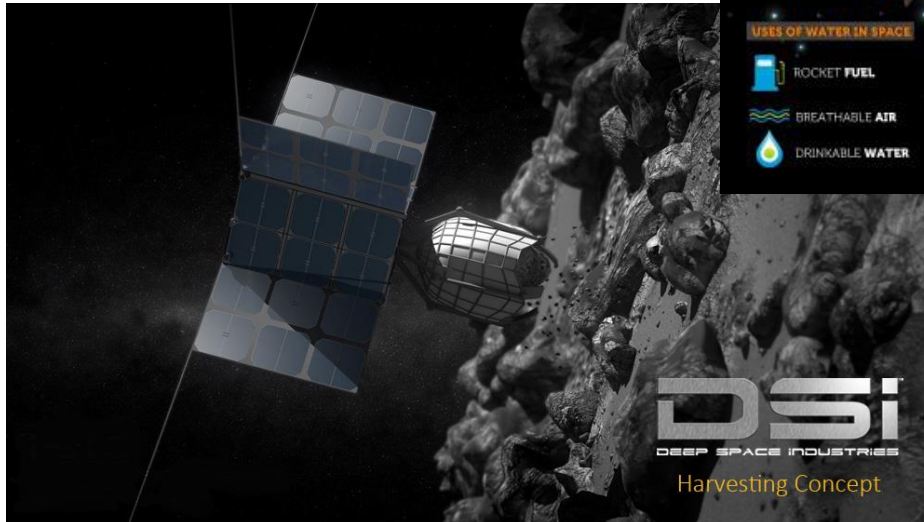
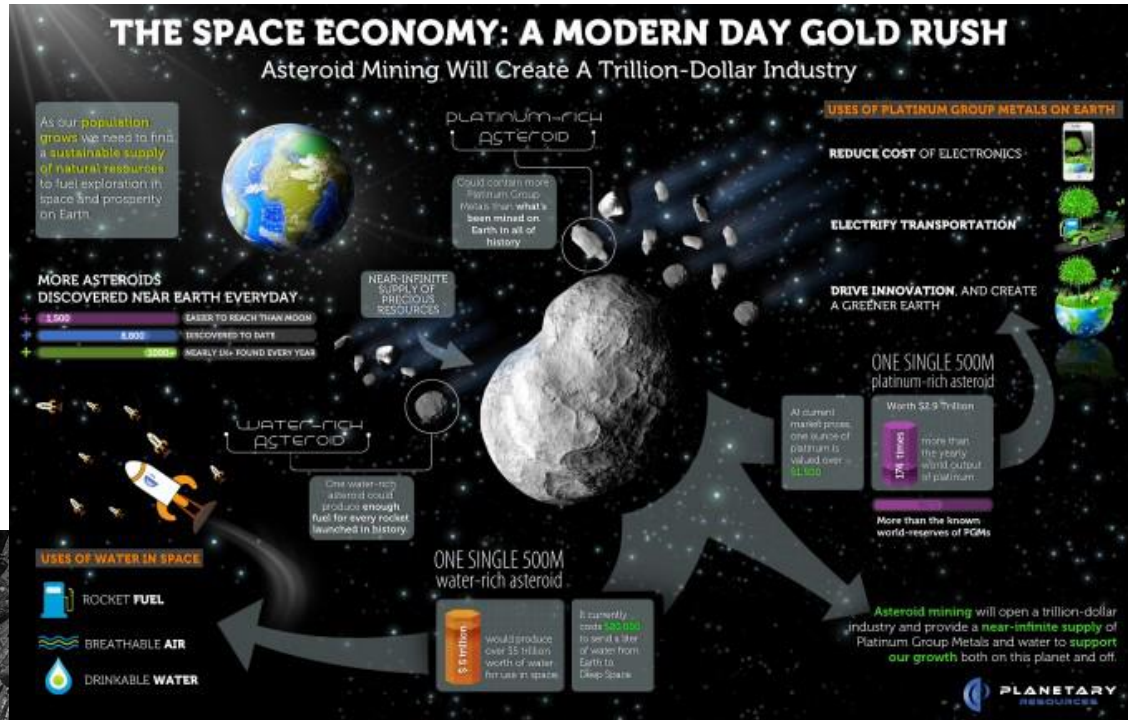
Science



- Better understanding of the early solar system.
- Fuel, oxygen and water storehouses
- Early trace of chemical components of life

Small bodies exploration, why?

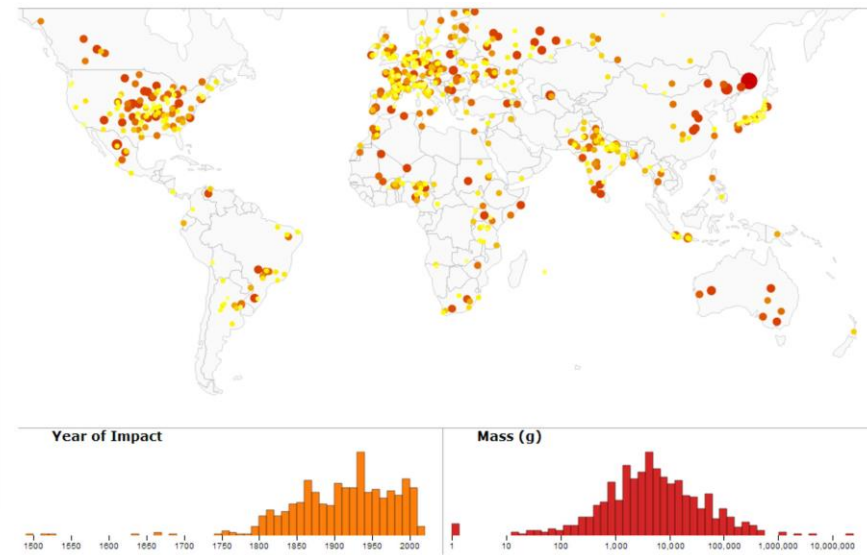
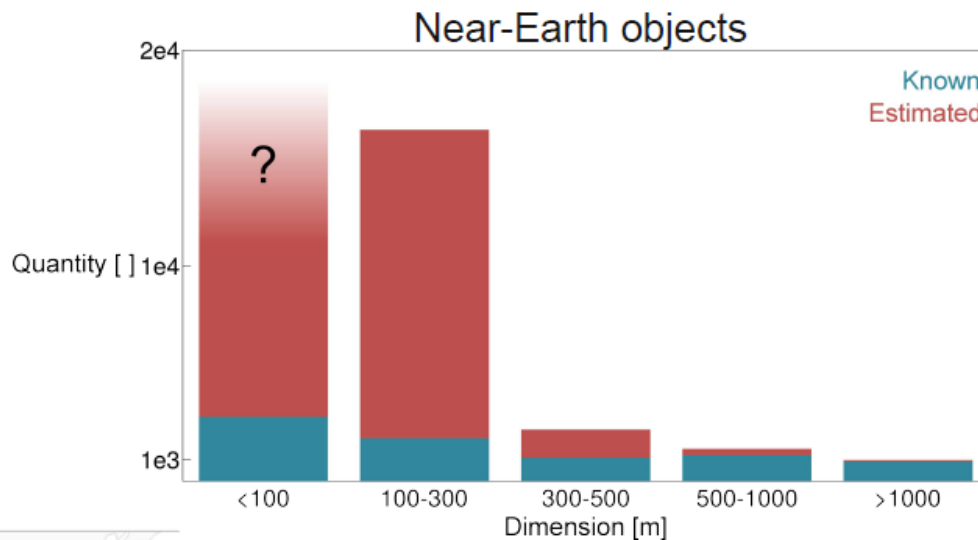
Economical



Small bodies exploration, why?

Defence

- There are less large asteroids than small ones
- It is important to monitor their positions for impacts



Small bodies exploration, why?

Defense



10m impacting body
1/50 years



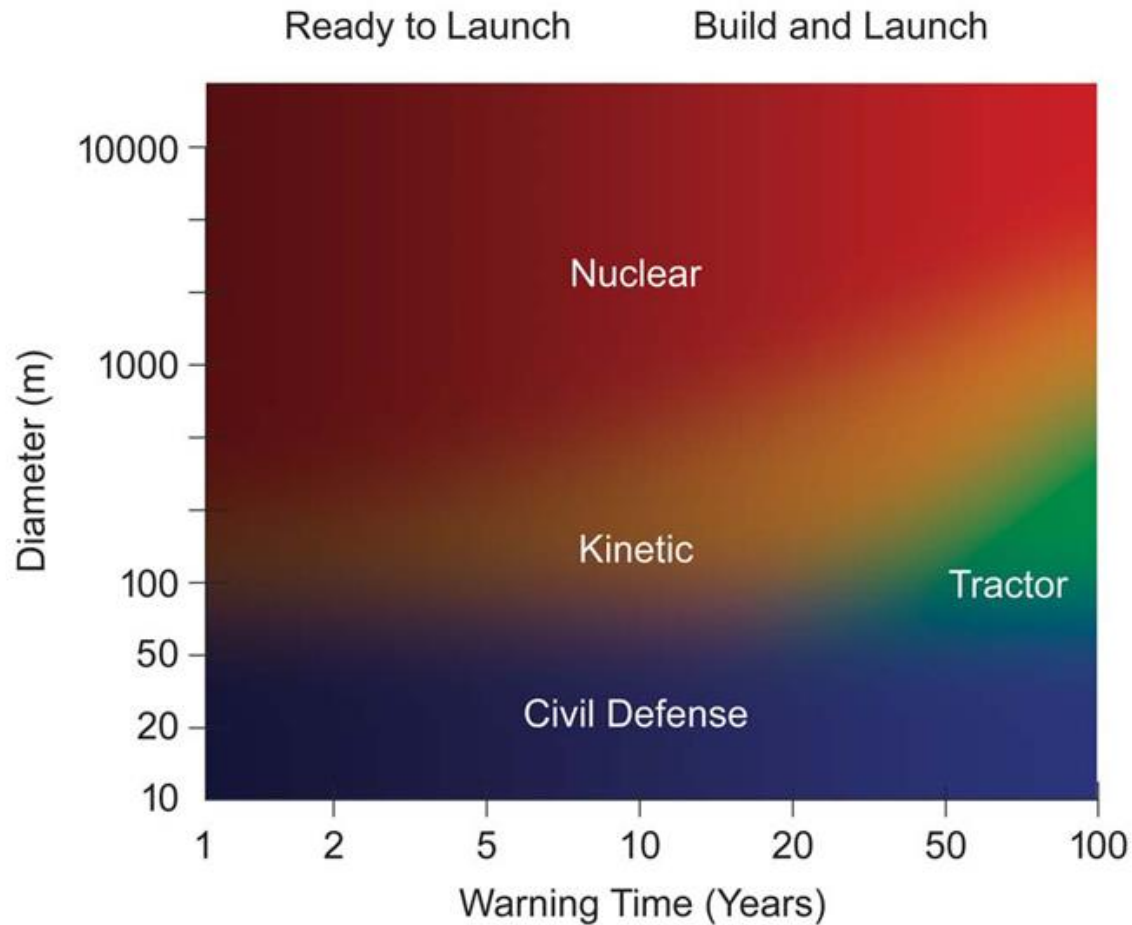
50m impacting body
1/100 years



10km impacting body
1/100 million years

Small bodies exploration, why?

Defense

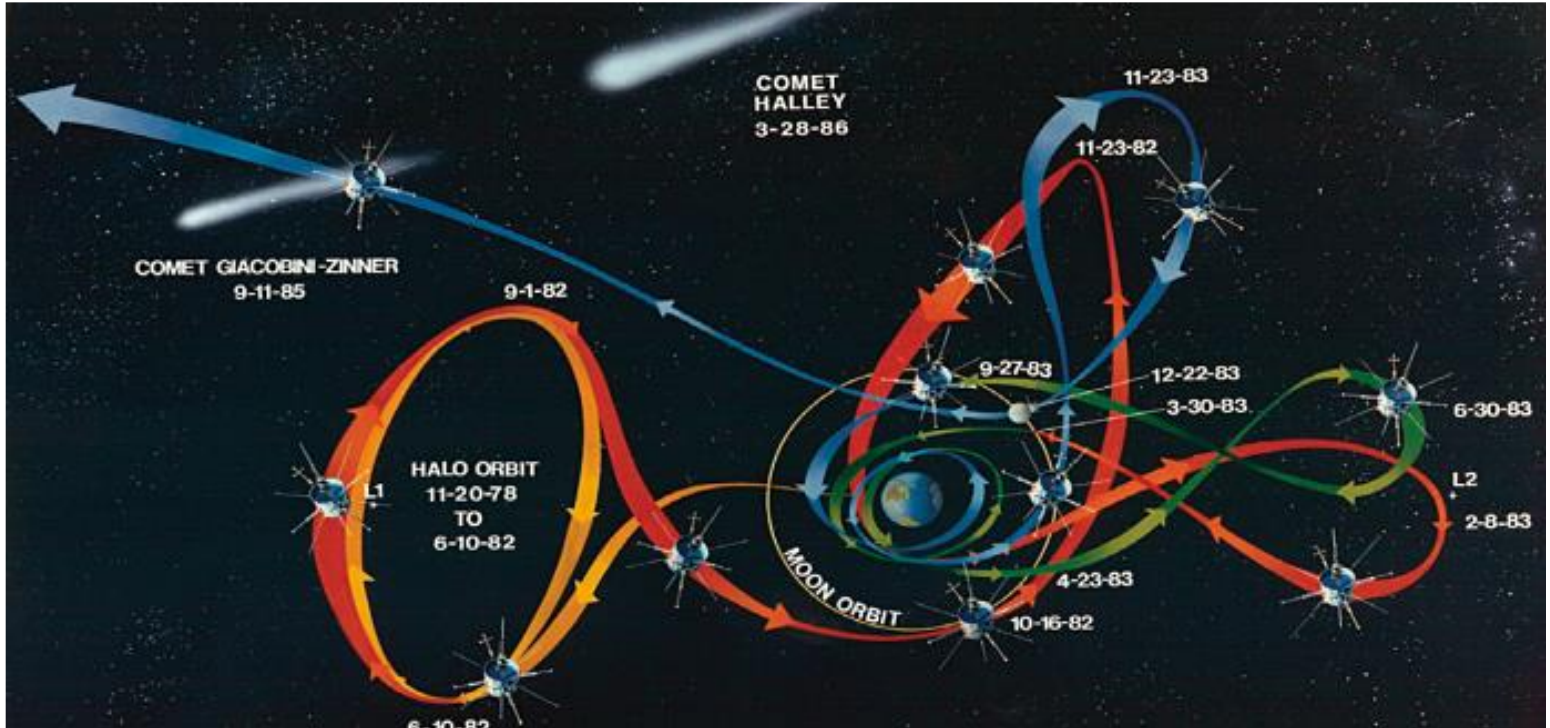


Milestones of small bodies exploration

- 1985 Fly-by of a comet (ISEE-3)
- 1991 Fly-by of an asteroid (Galileo)
- 1998 Orbit around the asteroid 433 Eros (NEAR Shoemaker)
- 2001 Landing on 433 Eros (NEAR Shoemaker)
- 2006 Sample return from a comet (Stardust)
- 2010 Sample return to Earth (Hayabusa)
- 2014 Orbiting and soft landing on a comet (Rosetta & Philae)
- 2015 Most distant object visited (New Horizons)
- 2016 Sample return from Bennu (Osiris-Rex)
- 2022 Asteroid Deflection (Dart)
- 2023 Orbit around Psyche in the main belt (Psyche)

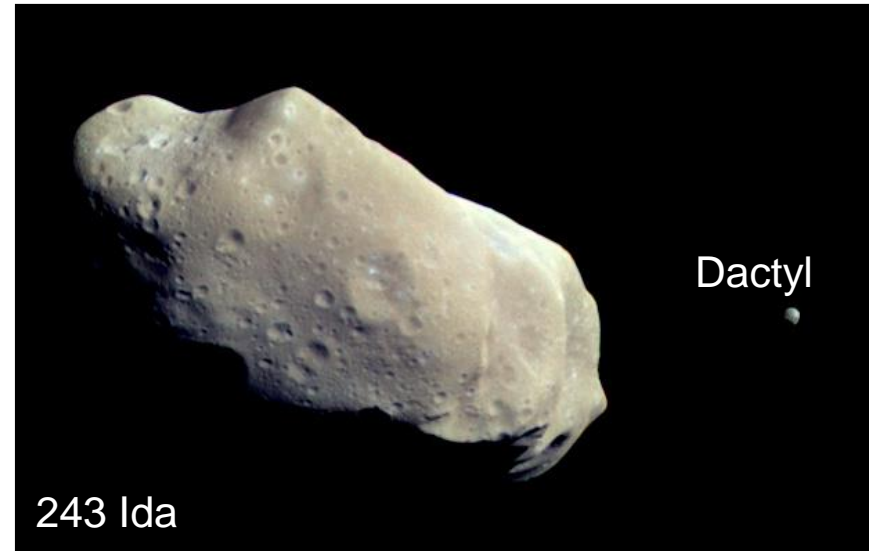
International Sun-Earth Explorer-3 (1978-1997)

1985



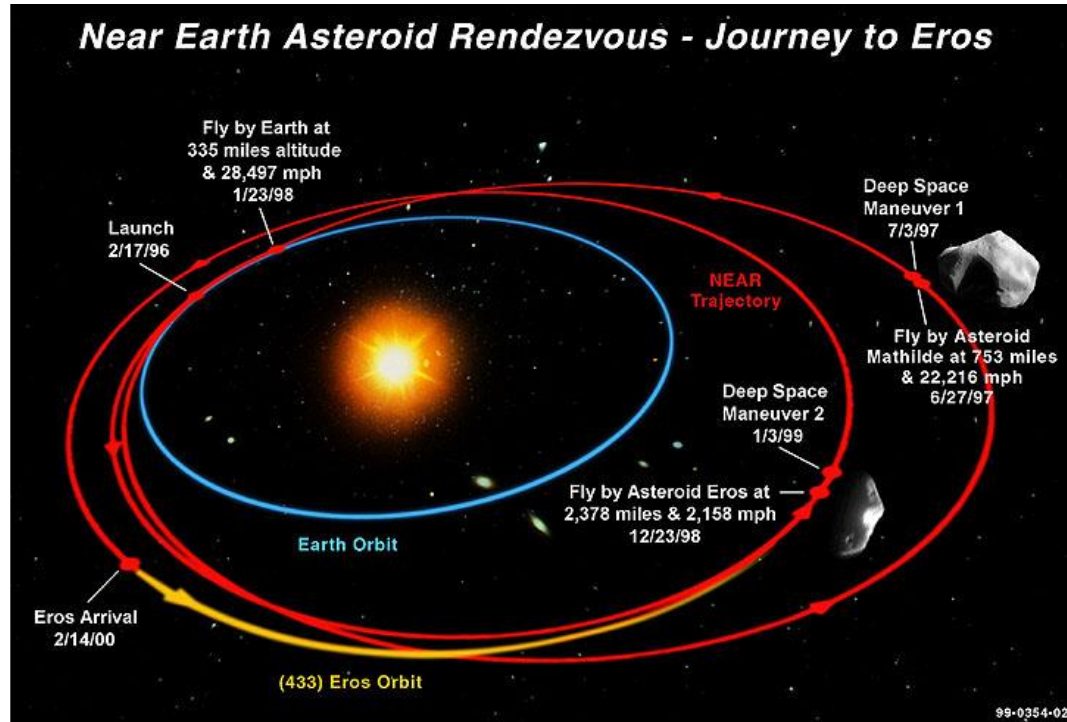
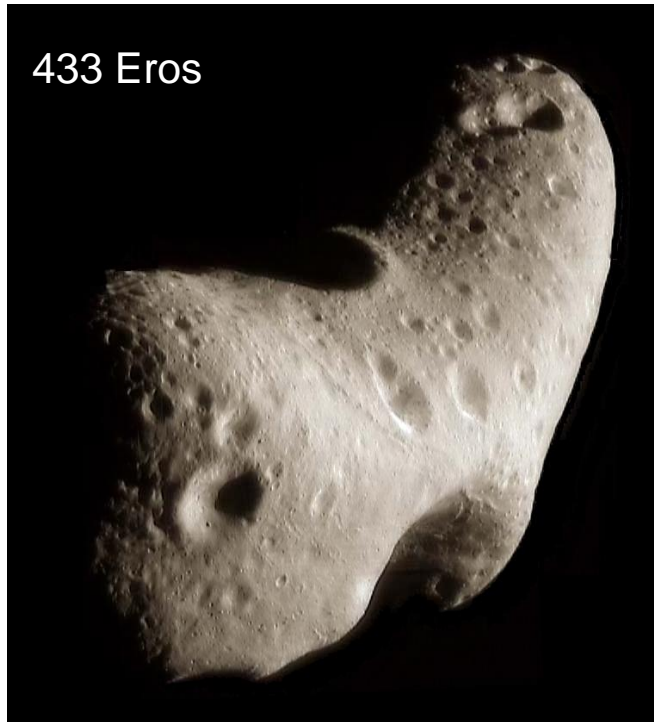
Galileo (1989-2003)

1991



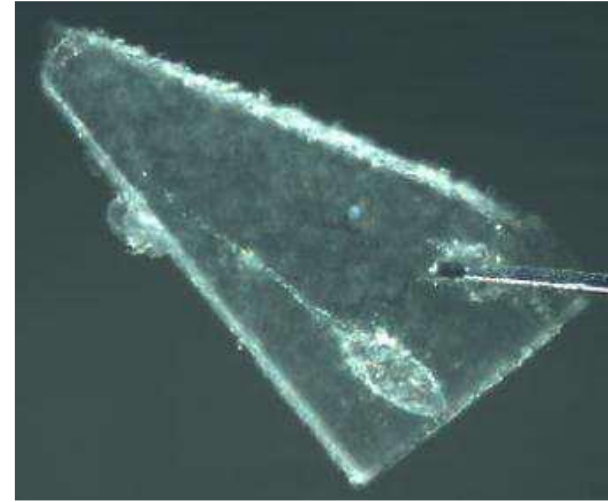
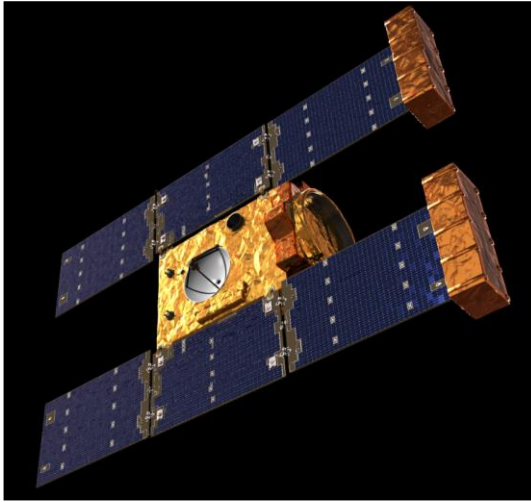
NEAR Shoemaker (1996-2001)

1998



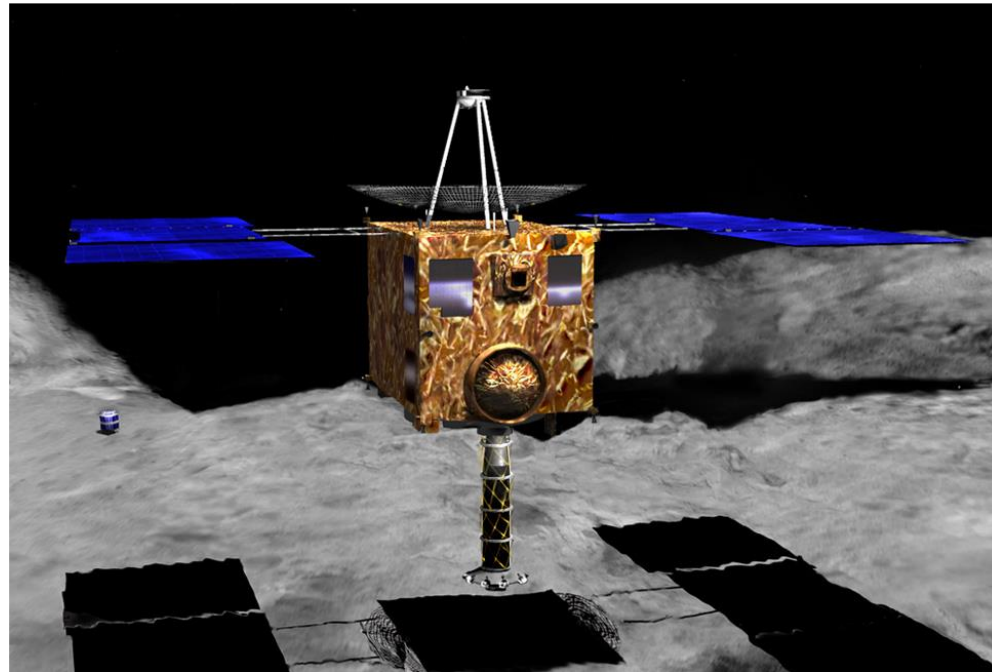
Stardust (1999-2006)

2001



Hayabusa (2003-2010)

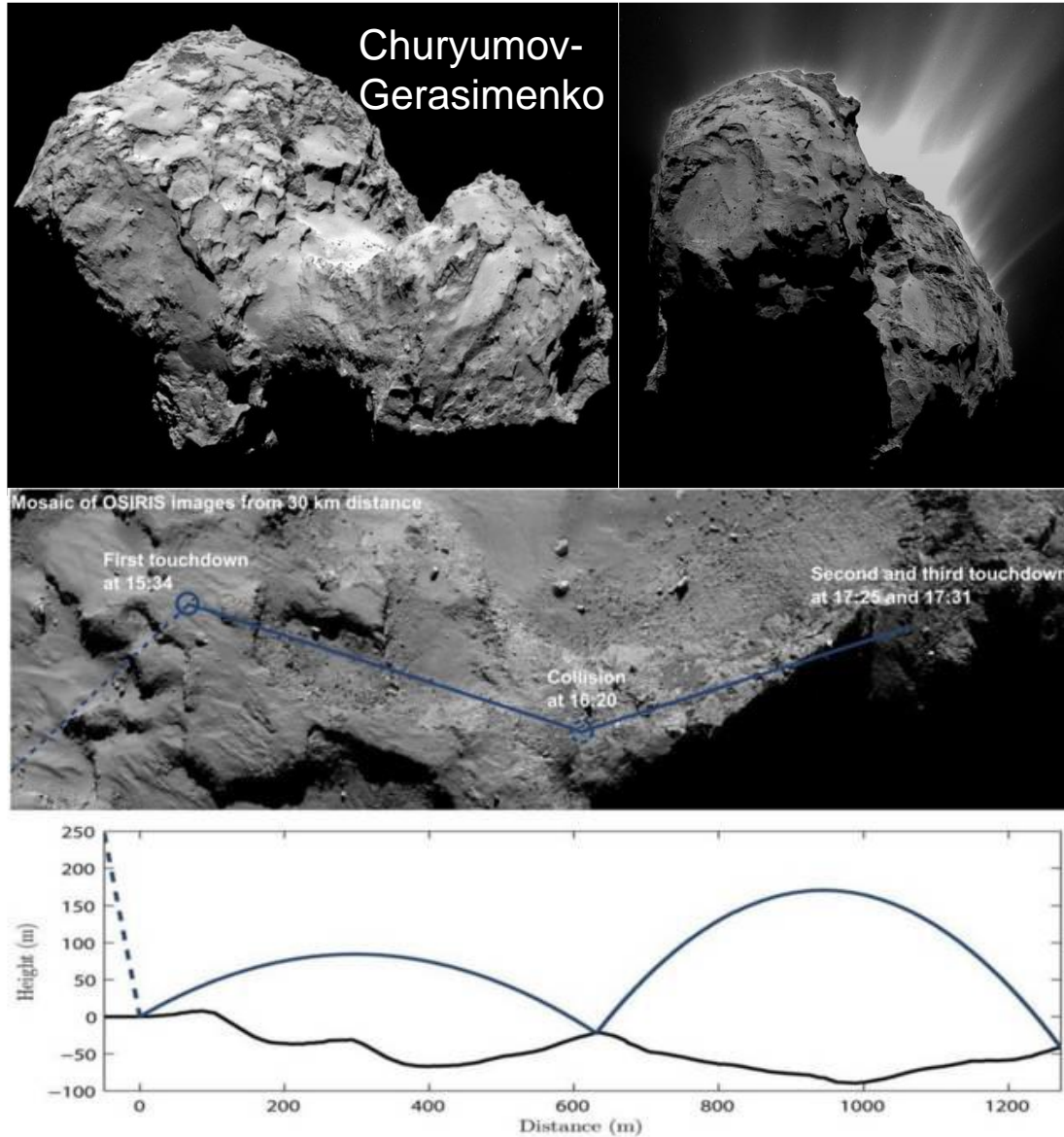
2006



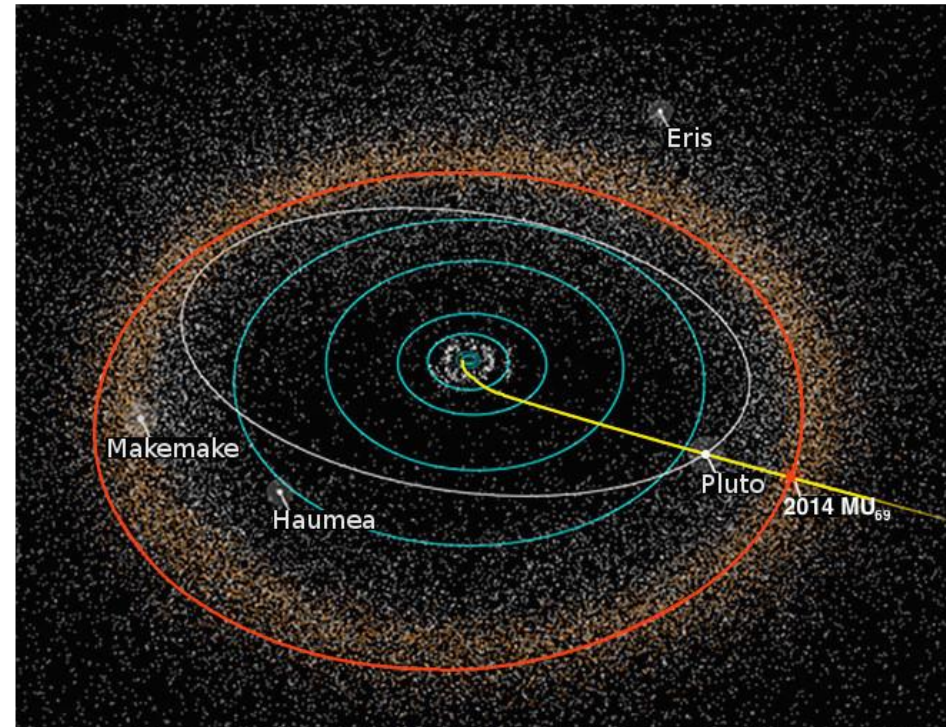
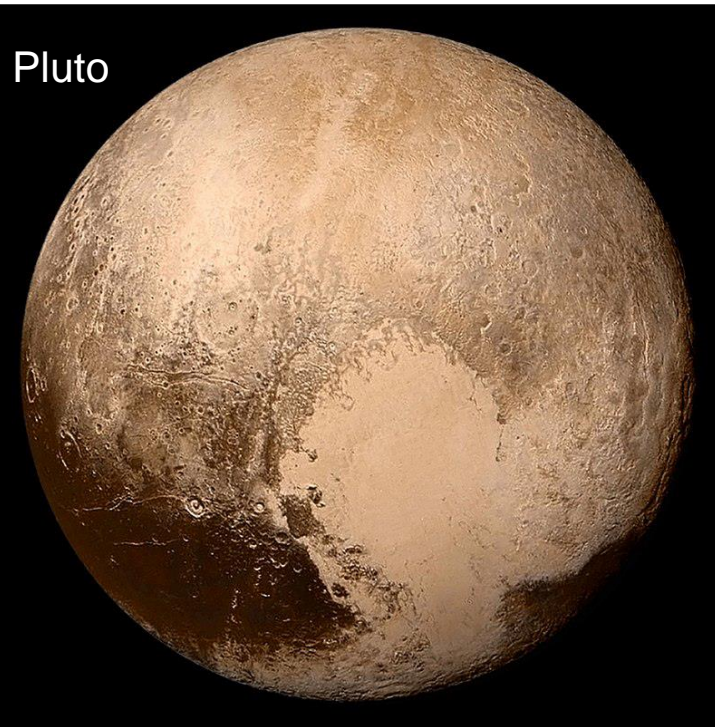
25143 Itokawa

Rosetta (2004-2016)

2010

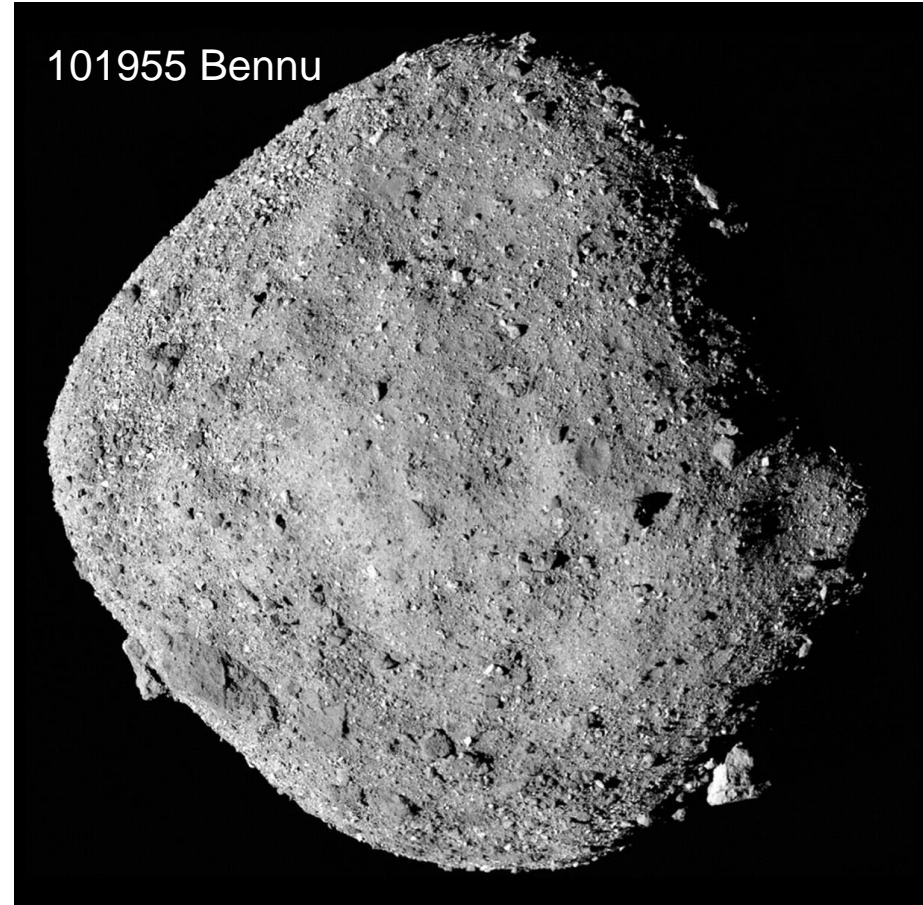


New Horizons (2006-now)



2014

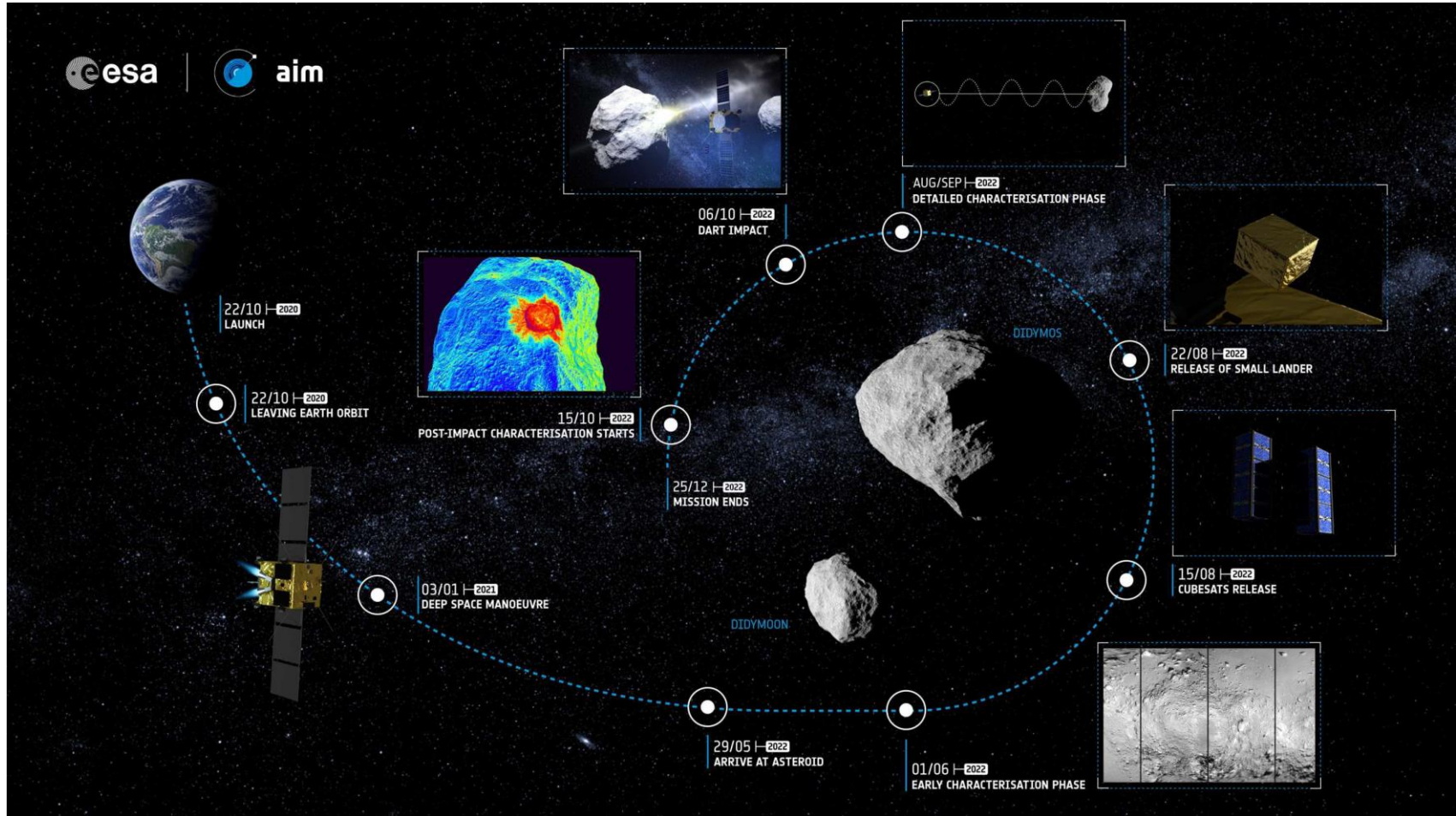
Osiris-REX (2016 – 2023)



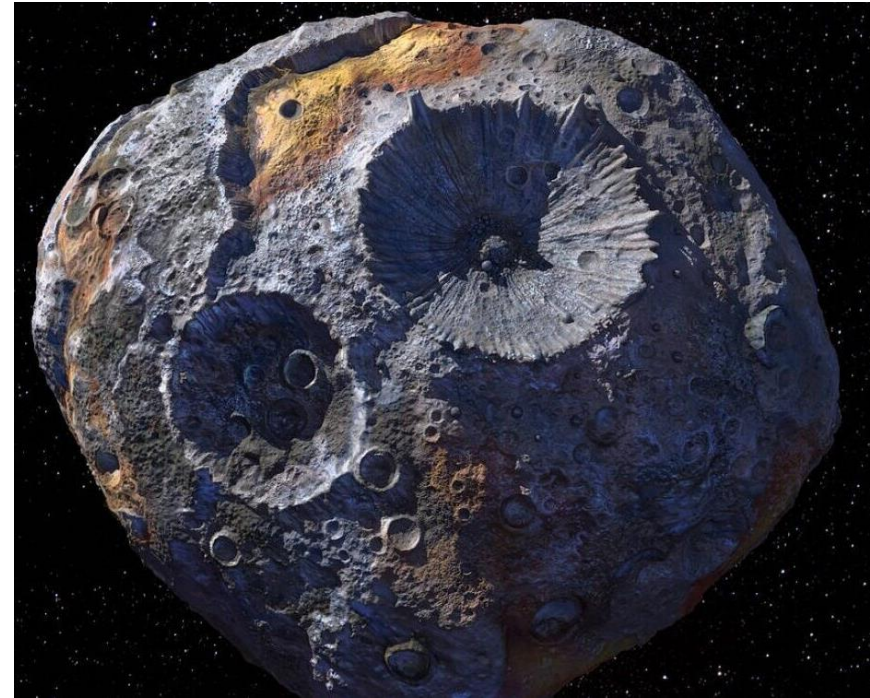
2016

DART (2022-2024)

2015



Psyche (2023-...)



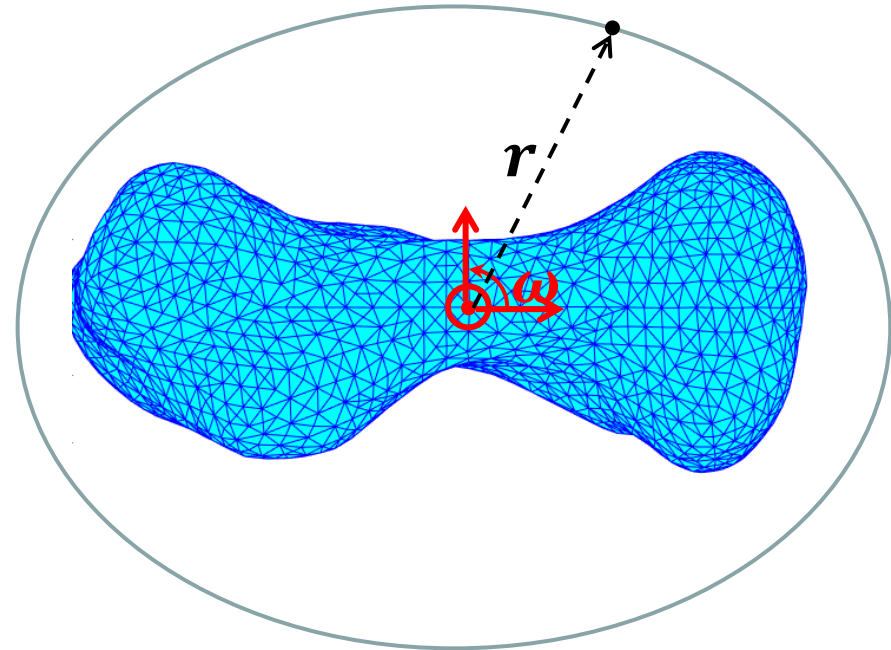
2022

Orbiting an irregular body

The position of a mass-less point, r , with respect to the center of mass of an irregular body in space depends of 2 factors.

The angular velocity of the body around its main inertia axis.

And the gravity potential of the body which is directly linked to its mass.



Equations of motion

$$\ddot{\mathbf{x}} + 2\boldsymbol{\omega}_A \times \dot{\mathbf{x}} + \boldsymbol{\omega}_A \times (\boldsymbol{\omega}_A \times \mathbf{x}) + \nabla U(\mathbf{x}) = \mathbf{0}$$

$\ddot{\mathbf{x}}$ = the acceleration

$2\boldsymbol{\omega}_A \times \dot{\mathbf{x}}$ = Coriolis force

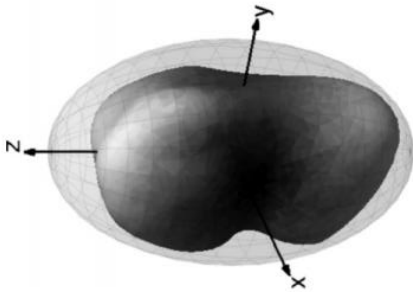
$\boldsymbol{\omega}_A \times (\boldsymbol{\omega}_A \times \mathbf{x})$ = centrifugal force

$\nabla U(\mathbf{x})$ = the gradient of the gravity potential

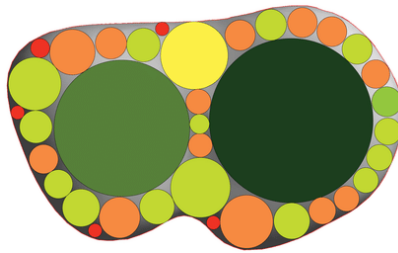
Gradient of the potential

$$\ddot{\mathbf{x}} + 2\boldsymbol{\omega}_A \times \dot{\mathbf{x}} + \boldsymbol{\omega}_A \times (\boldsymbol{\omega}_A \times \mathbf{x}) + \nabla U(\mathbf{x}) = \mathbf{0}$$

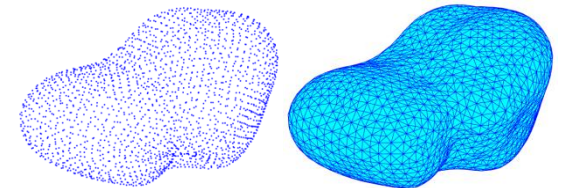
Spherical
harmonic model



Mass
concentration

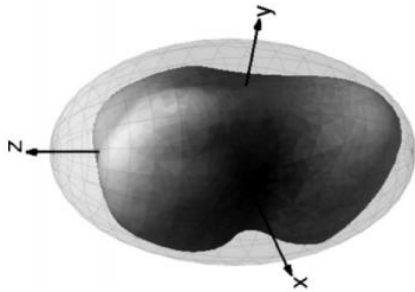


Polyhedrons



Spherical harmonic

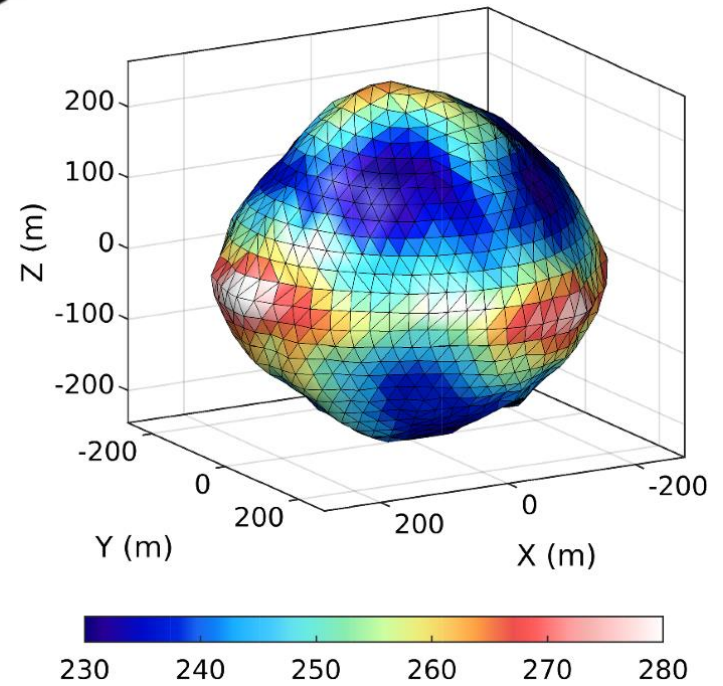
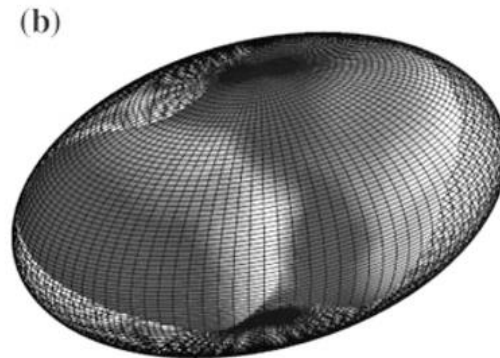
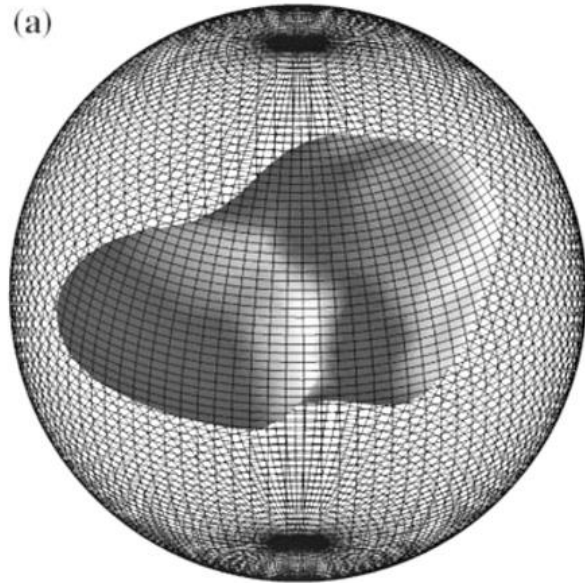
Spherical
harmonic
model



Spherical harmonics are straightforward and not computationally demanding but not very accurate for irregular bodies.

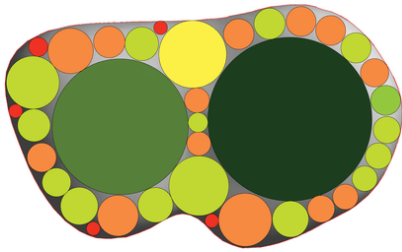
$$U = \frac{GM}{R} \sum_{n=0}^{\infty} \sum_{m=0}^n \left(\frac{R}{r}\right)^{n+1} P_{nm}(\sin \theta) \begin{bmatrix} \cos(m\lambda) \\ \sin(m\lambda) \end{bmatrix} \cdot \begin{bmatrix} C_{nm} \\ S_{nm} \end{bmatrix}$$

Spherical harmonic



Mascons

Mass
concentration



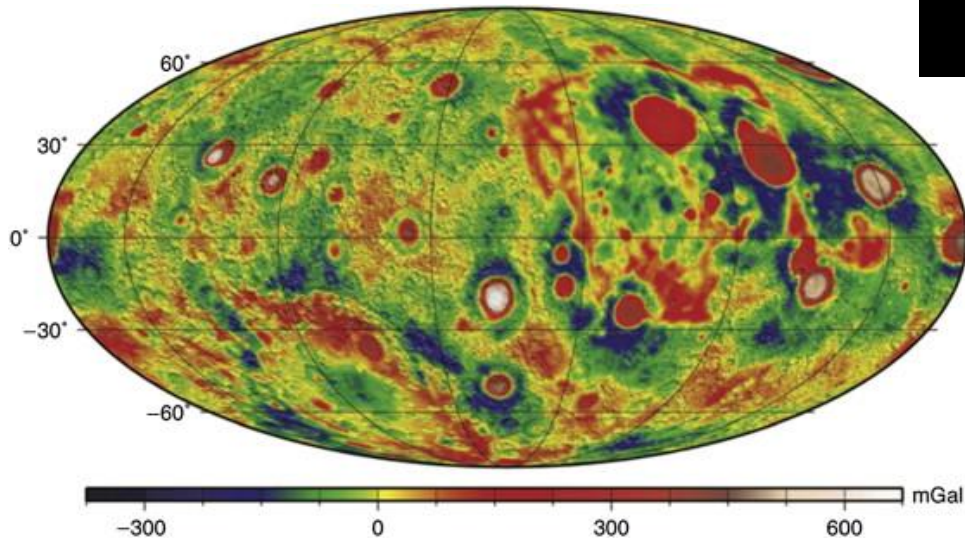
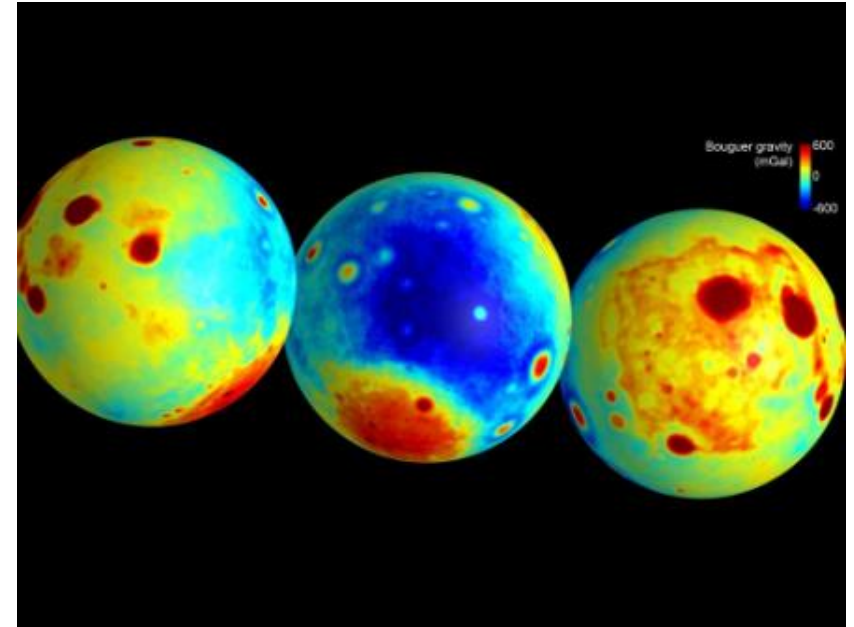
Simple representation but allows to represent the internal density distribution. The resolution depends on the number of masses considered.

$$U = G \sum_{i=0}^N \frac{m_i}{|\mathbf{r} - \mathbf{r}_i|}$$

Moon's Lumpy Gravity

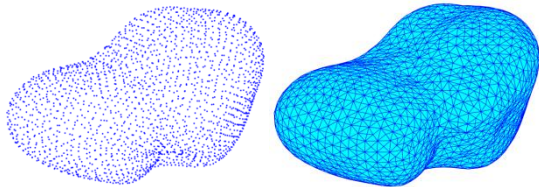
Mascons were discovered during the planning for Apollo moon missions.

They are the results of asteroids impacts. The heat melt the moon rocks into denser parts.



Polyhedrons

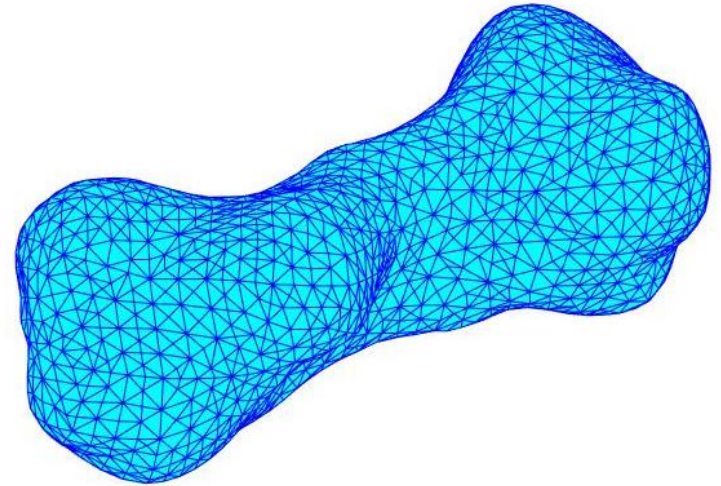
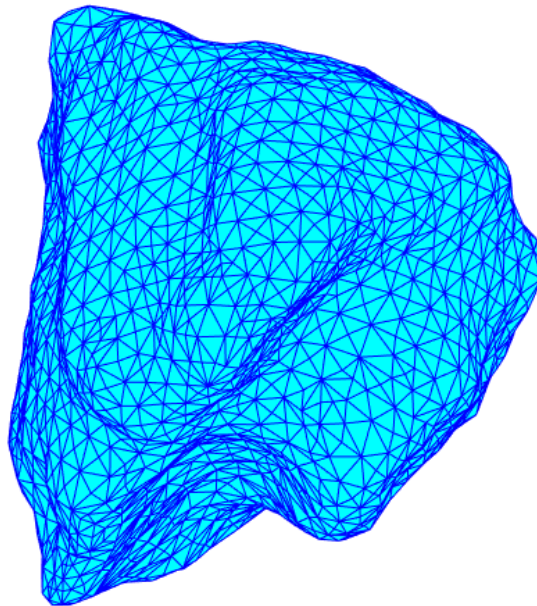
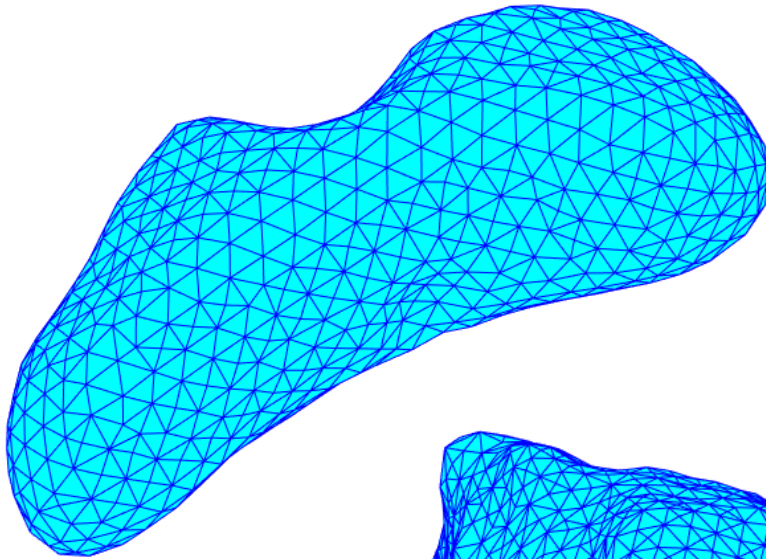
Polyhedrons



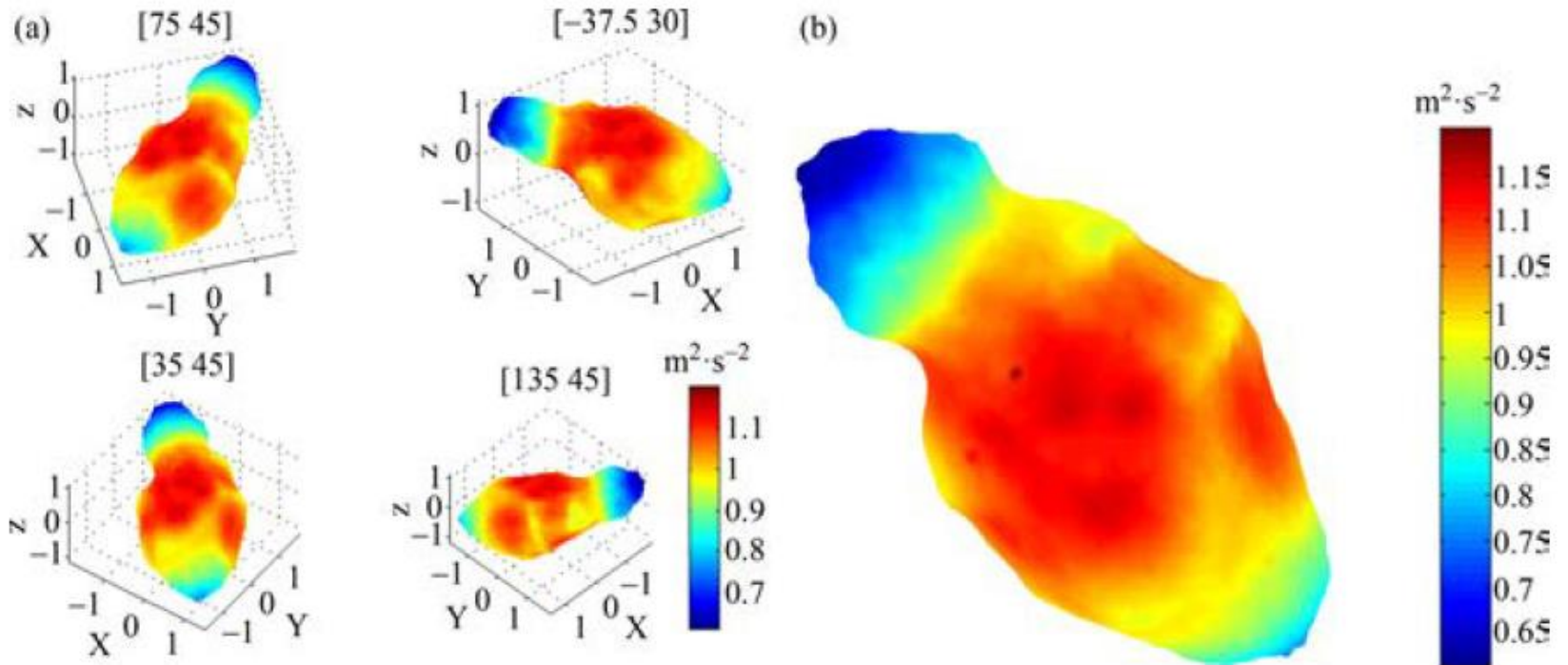
Analytical expression for the potential of an arbitrary polyhedron. Highly accurate but computationally intensive.

$$U = \frac{1}{2} G \rho \sum_e \mathbf{r}_e \cdot \mathbf{E}_e \cdot \mathbf{r}_e \cdot L_e - \frac{1}{2} G \rho f \sum_f \mathbf{r}_f \cdot \mathbf{F}_f \cdot \mathbf{r}_f \cdot \omega_f$$

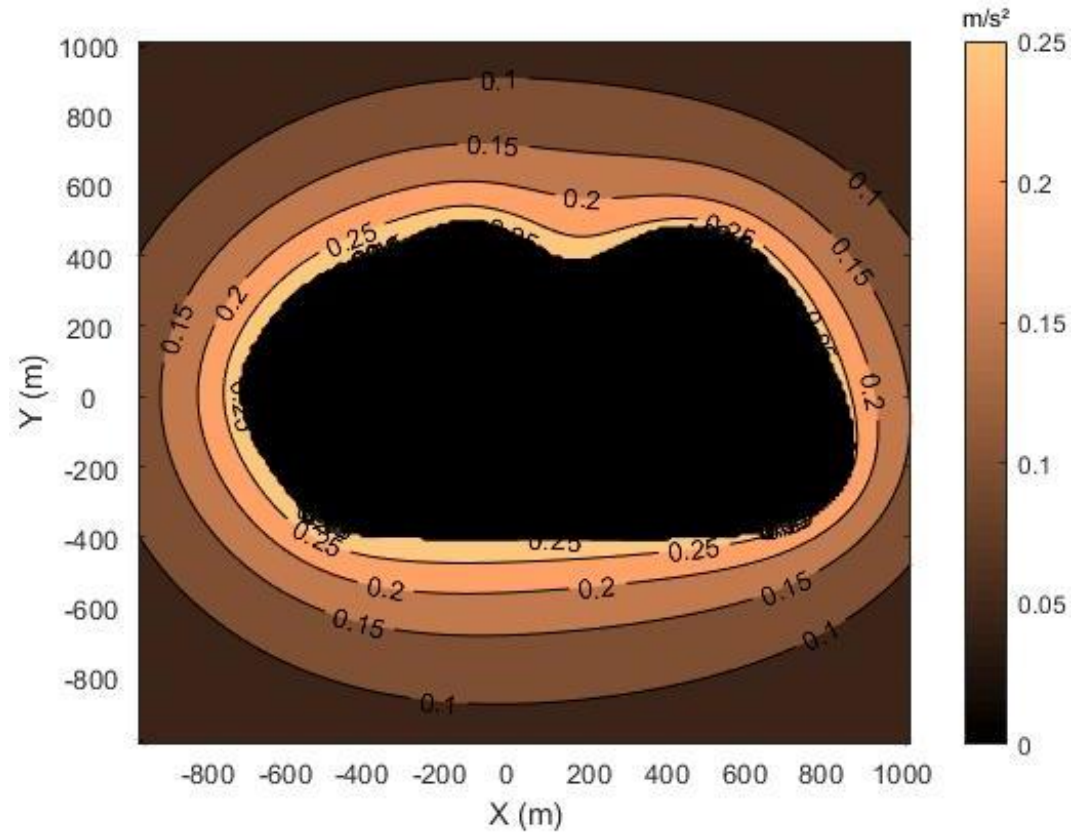
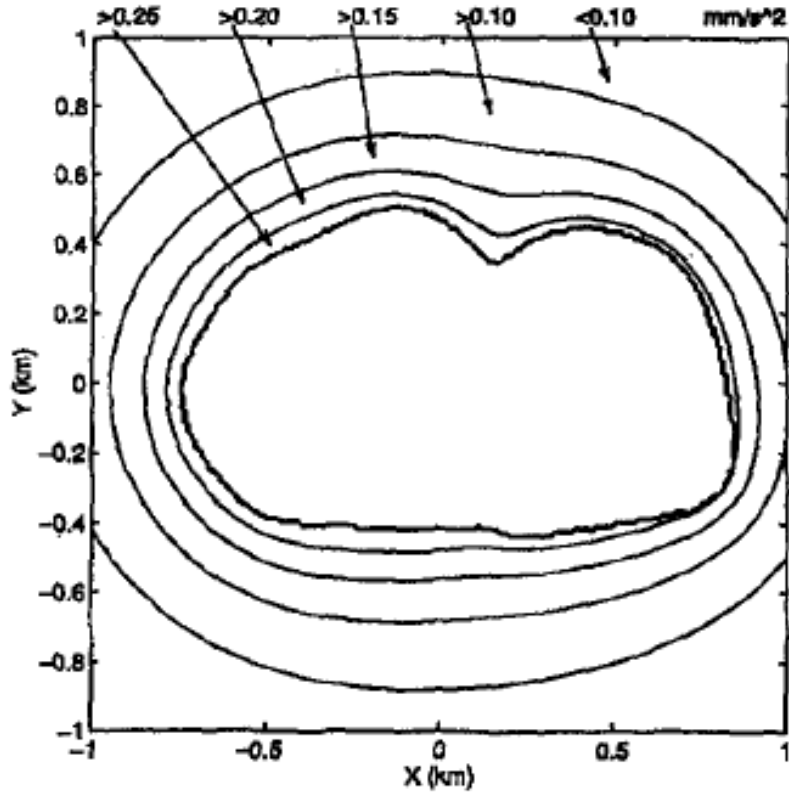
All kind of meshes



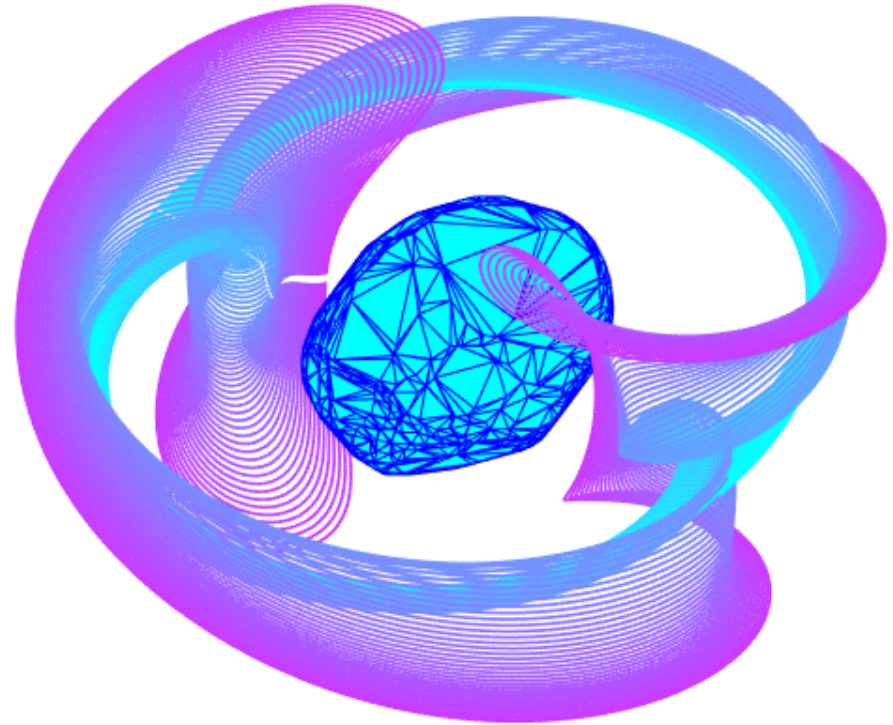
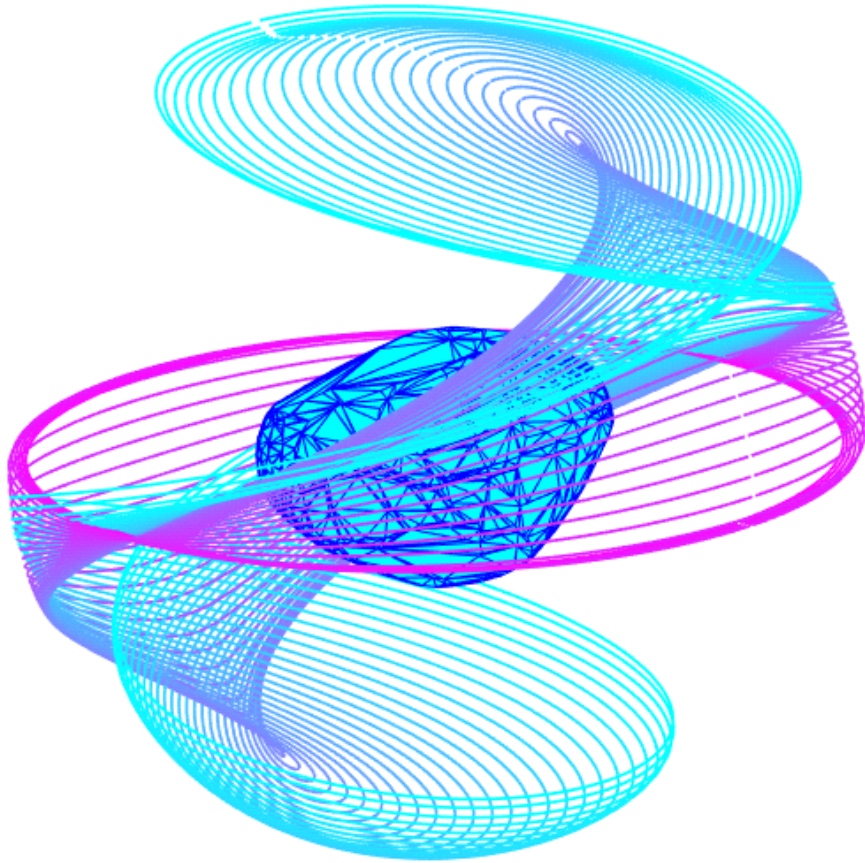
Surface potential



Gravity potential



Orbits shapes



Conclusions



Missions to small bodies are complex and challenging

Complex shape and weak gravity strongly perturb asteroid's dynamical environment

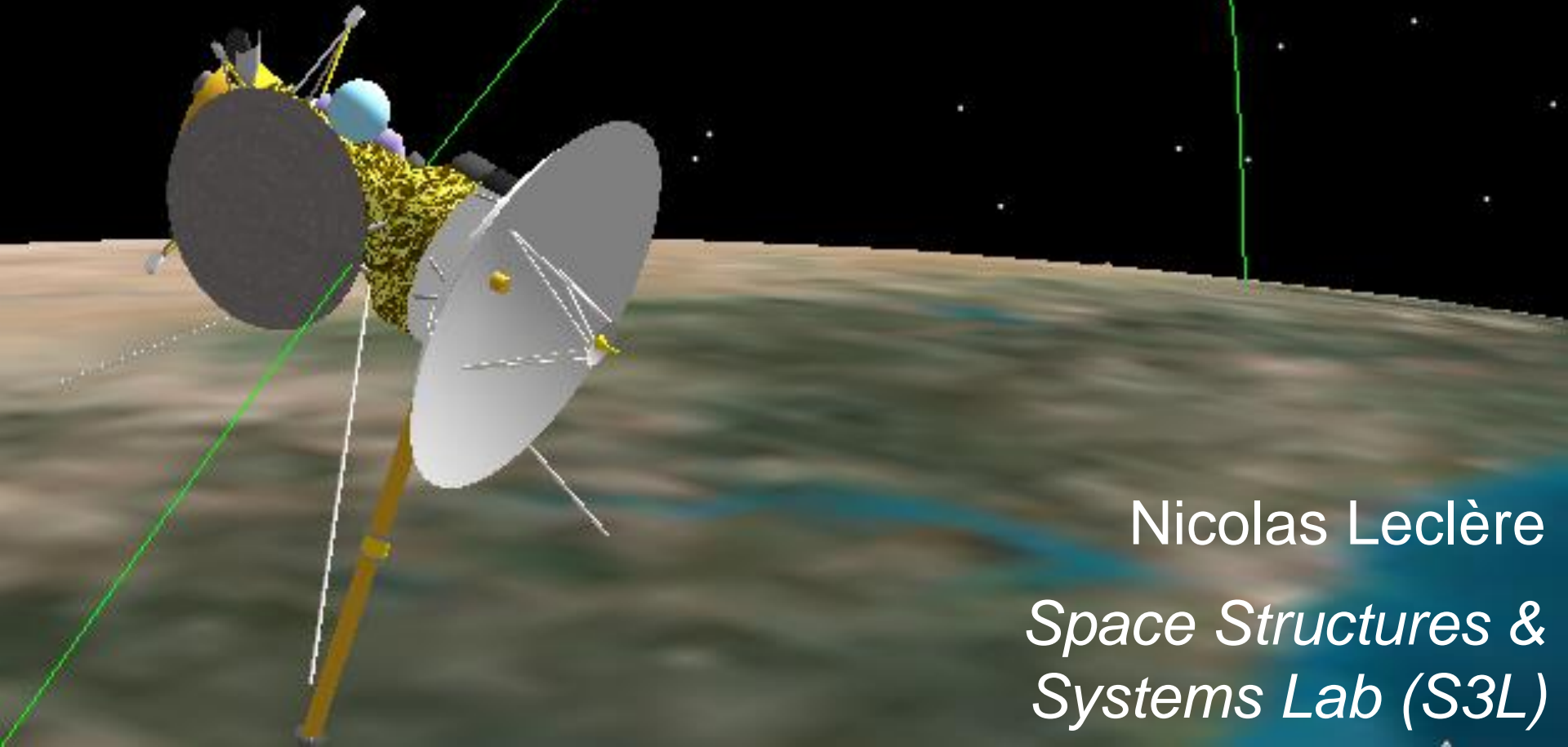
Growing interest for asteroid missions and increasing ambitions for the future plans

Cassini Classical Orbit Elements
Time (UTCG): 15 Oct 1997 09:18:54.000
Semi-major Axis (km): 6685.637000
Eccentricity: 0.020566
Inclination (deg): 30.000
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Aerodynamics

(AERO0024)

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