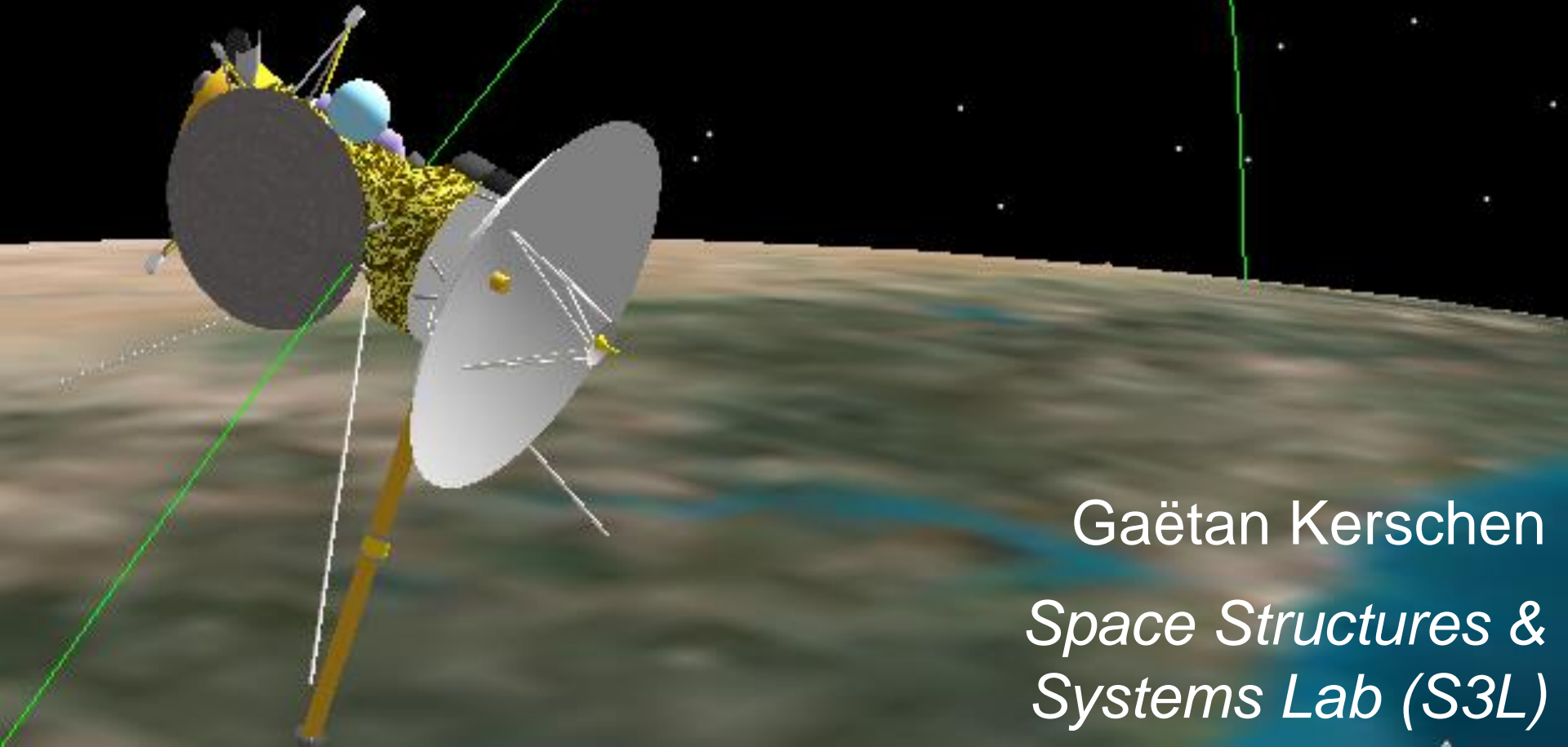


Cassini Classical Orbit Elements  
Time (UTCG): 15 Oct 1997 09:18:54.000  
Semi-major Axis (km): 6685.637000  
Eccentricity: 0.020566  
Inclination (deg): 30.000  
RAAN (deg): 150.546  
Arg of Perigee (deg): 230.000  
True Anomaly (deg): 136.530  
Mean Anomaly (deg): 134.891

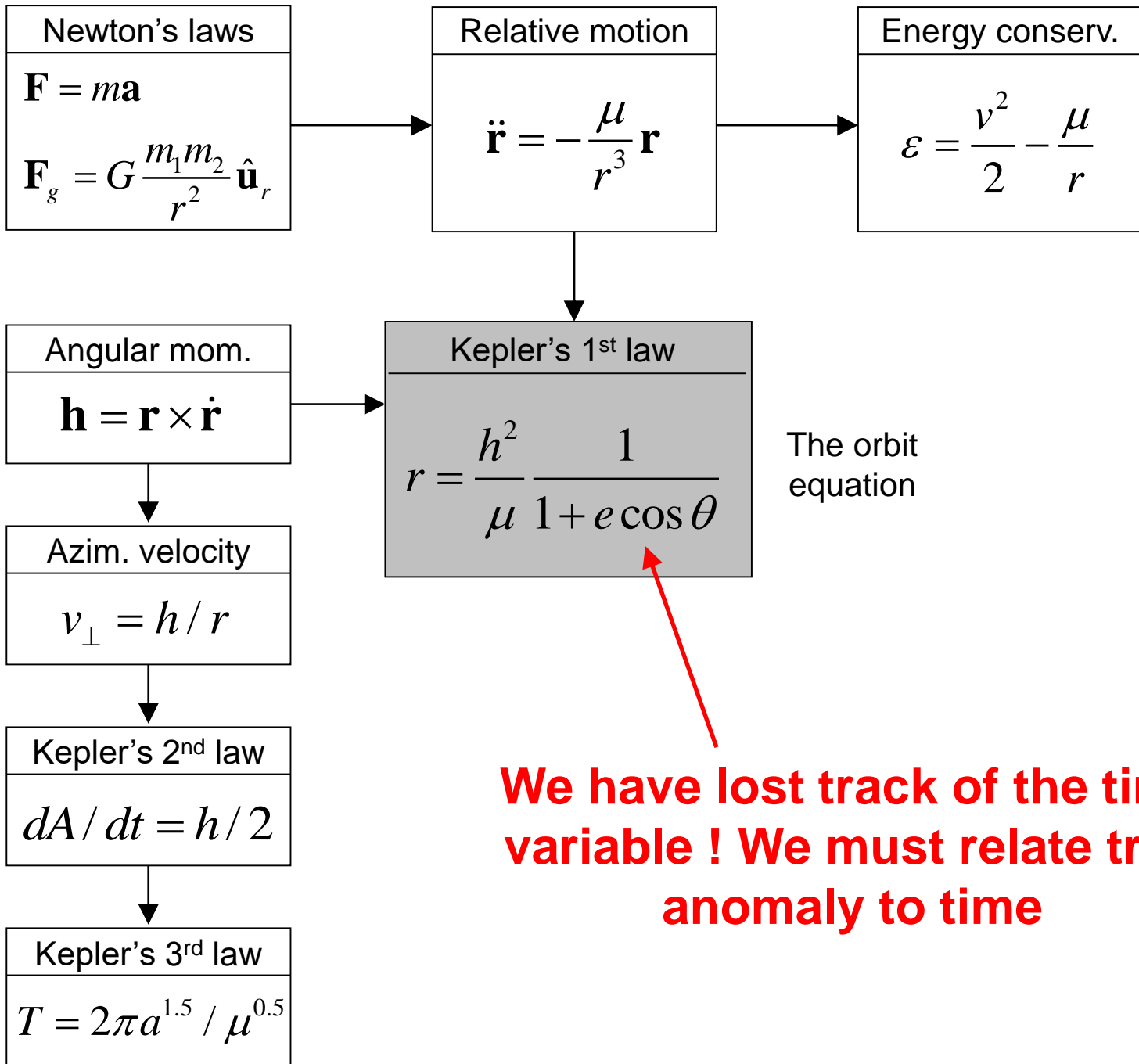
# Aerodynamics

(AERO0024)

## 3. *The Orbit in Time*



Gaëtan Kerschen  
*Space Structures &  
Systems Lab (S3L)*



# Importance/Complexity of Time Measurement

Applications such as GPS rely on an extremely precise time measurement system:

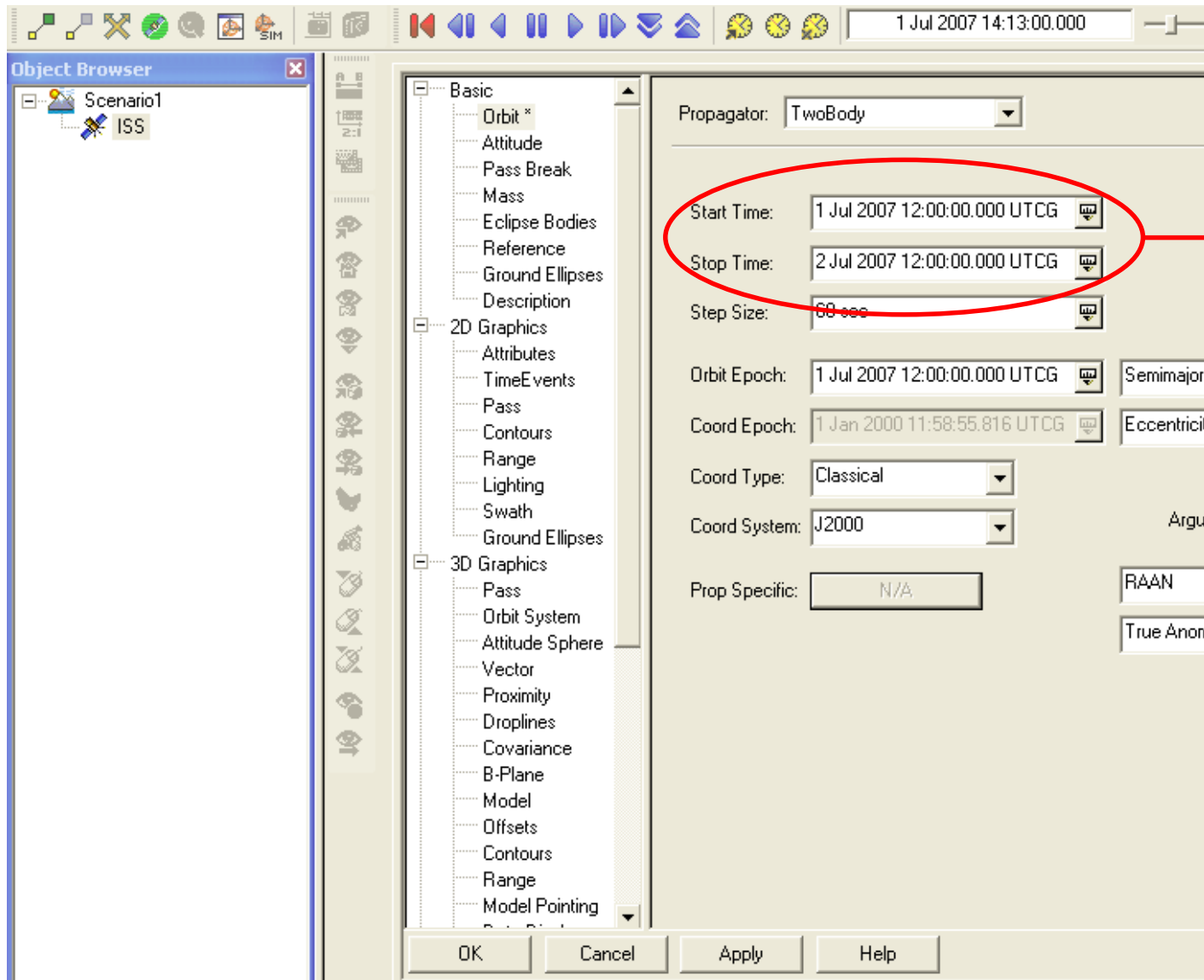
an error of 1 nanosecond translates into an error of 30cm in the distance.

What time is it? Well, no one knows for sure

[guardian.co.uk](http://guardian.co.uk)

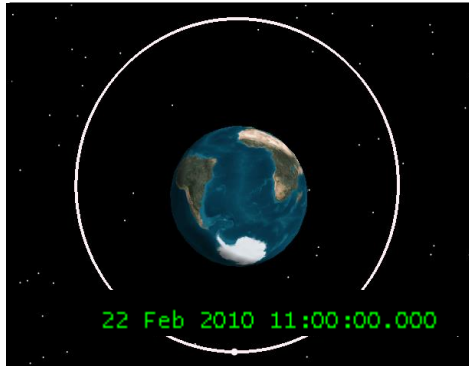
As the Earth spins slower, methods of telling time diverge. Experts warn this could end in disaster

# Complexity: STK



- EpSec
- EpMin
- EpHr
- EpDay
- LCLG
- UTCG
- LCLJ
- UTCJ
- YYDDD
- UTCJFOUR
- JDate
- YYYYDDD
- JDateOff
- MisElap
- YYYYMMDD
- YYYY/MM/DD
- GMT
- ModJDate
- JED
- TDTG
- TAIG
- TAIJ
- GPSG
- GPS
- DD/MM/YYYY
- EarthEpTU
- SunEpTU
- YYYY:MM:DD
- GPSZ
- TDBG
- Format...
- Update Animation Time

# The Orbit in Time



Orbital position as a function of time



Time systems

# Time Since Periapsis

What is the time required to fly between any two true anomaly?

$$\frac{dA}{dt} = \frac{1}{2} r^2 \frac{d\theta}{dt} = \frac{h}{2} = \text{constant}$$

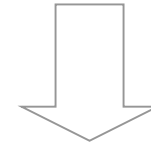
Kepler's second law

$$r = \frac{h^2}{\mu} \frac{1}{1 + e \cos \theta}$$

Orbit equation

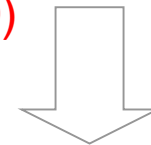


$$\frac{\mu^2}{h^3} dt = \frac{d\theta}{(1 + e \cos \theta)^2}$$



$$\frac{\mu^2}{h^3} (t - t_p) = \int_0^\theta \frac{d\theta}{(1 + e \cos \theta)^2}$$

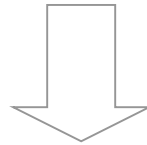
Sixth missing constant ( $t_p=0$ )



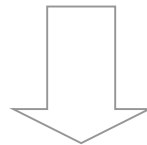
$$\frac{\mu^2}{h^3} t = \int_0^\theta \frac{d\theta}{(1 + e \cos \theta)^2}$$

# Circular Orbits

$$\frac{\mu^2}{h^3} t = \int_0^\theta \frac{d\theta}{(1 + e \cos \theta)^2}$$



$$\frac{\mu^2}{h^3} t = \int_0^\theta d\theta$$



$$t = \frac{h^3}{\mu^2} \theta = \frac{r^{3/2}}{\sqrt{\mu}} \theta = \frac{T}{2\pi} \theta \quad \rightarrow \theta = \frac{2\pi}{T} t$$

Obvious, because the angular velocity is constant.

# Elliptic Orbits

Because the angular velocity of a spacecraft along an eccentric orbit is continuously varying, the expression of the angular position versus time is no longer trivial.

$$\frac{\mu^2}{h^3} t = \int_0^\theta \frac{d\theta}{(1 + e \cos \theta)^2}$$

Mean anomaly,  $M$

$$= \frac{1}{(1 - e^2)^{3/2}} \left[ 2 \tan^{-1} \left( \sqrt{\frac{1 - e}{1 + e}} \tan \frac{\theta}{2} \right) - \frac{e \sqrt{1 - e^2} \sin \theta}{1 + e \cos \theta} \right]$$

$$T_{\text{ellip}} = 2\pi \sqrt{\frac{a^3}{\mu}}$$

$$h = \sqrt{\mu a (1 - e^2)}$$

$$\Rightarrow M = \frac{\mu^2}{h^3} (1 - e^2)^{3/2} t = \frac{2\pi}{T} t = nt$$

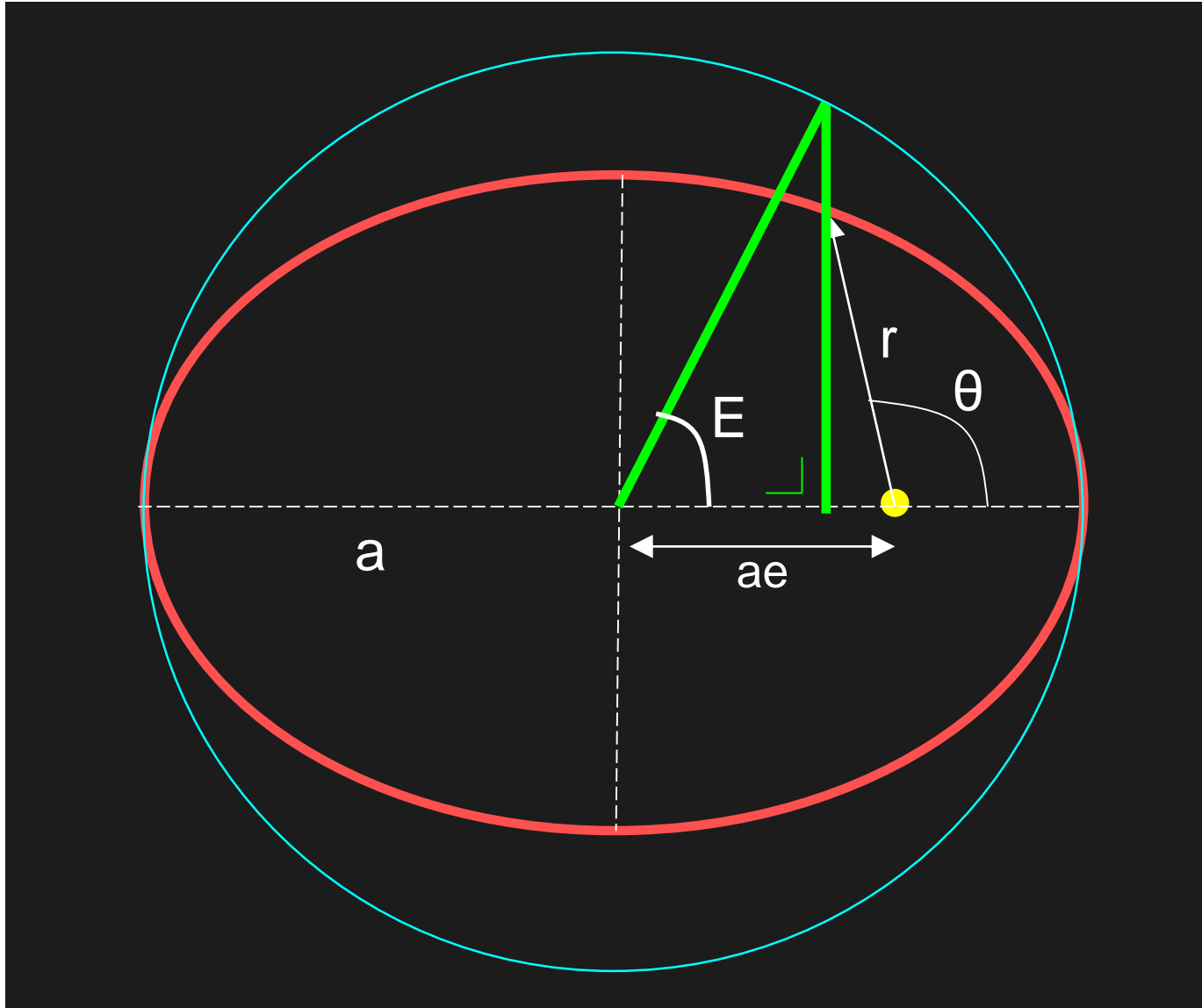


# Mean Anomaly Is Related to Time

For circular orbits, the mean  $M$  and true anomalies  $\theta$  are identical.

For elliptic orbits, the mean anomaly represents the angular displacement of a fictitious body moving around the ellipse at the constant angular speed  $n$ .

# Eccentric Anomaly Is Related to Position



# Eccentric Anomaly: Relation with Mean Anomaly ?

$$a \cos E = ae + r \cos \theta \text{ Graph}$$

$$r = a \frac{1 - e^2}{1 + e \cos \theta} \quad \text{L2}$$



$$\cos E = \frac{e + \cos \theta}{1 + e \cos \theta}$$



$$\cos^2 E + \sin^2 E = 1$$

$$\sin E = \frac{\sqrt{1 - e^2} \sin \theta}{1 + e \cos \theta}$$

$$\cos E = \frac{e + \cos \theta}{1 + e \cos \theta}$$

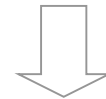
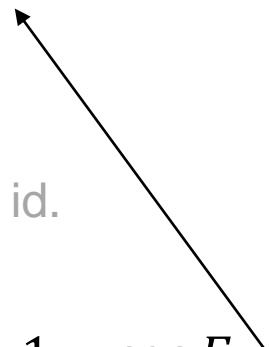


$$\sin E = \frac{\sqrt{1 - e^2} \sin \theta}{1 + e \cos \theta}$$

trig. id.



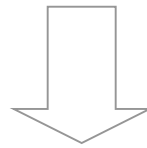
$$\tan^2 \frac{E}{2} = \frac{1 - \cos E}{1 + \cos E} = \frac{1 - e}{1 + e} \frac{1 - \cos \theta}{1 + \cos \theta} = \frac{1 - e}{1 + e} \tan^2 \frac{\theta}{2}$$



$$E = 2 \tan^{-1} \left( \sqrt{\frac{1 - e}{1 + e}} \tan \frac{\theta}{2} \right)$$

# Kepler's Equation

$$M = 2 \tan^{-1} \left( \sqrt{\frac{1-e}{1+e}} \tan \frac{\theta}{2} \right) - \frac{e \sqrt{1-e^2} \sin \theta}{1+e \cos \theta} \quad E = 2 \tan^{-1} \left( \sqrt{\frac{1-e}{1+e}} \tan \frac{\theta}{2} \right)$$

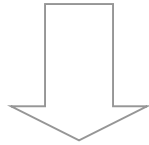


$$M = nt = E - e \sin E$$

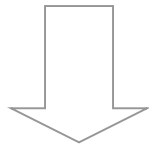
It relates time, in terms of  $M=nt$ , to position, in terms of  $E$ ,  $r=a(1-e \cos E)$ .

# Usefulness of Kepler's Equation

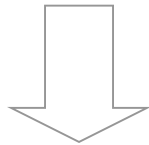
$\theta$  and orbit are given



$$E = 2 \tan^{-1} \left( \sqrt{\frac{1-e}{1+e}} \tan \frac{\theta}{2} \right)$$



$$M = E - e \sin E$$



$$t = \frac{M}{2\pi} T$$

Practical application:

Determine the time at which a satellite passes from sunlight into the Earth's shadow (the location of this point is known from the geometry).

# Example

A geocentric elliptic orbit has a perigee radius of 9600 km and an apogee radius of 21000 km. Calculate the time to fly from perigee to a true anomaly of  $120^\circ$ .

$$e = \frac{r_a - r_p}{r_a + r_p} = \frac{21000 - 9600}{21000 + 9600} = 0.37255$$

$$E = 2 \tan^{-1} \left( \sqrt{\frac{1-e}{1+e}} \tan \frac{\theta}{2} \right) = 1.7281 \text{ rad}$$

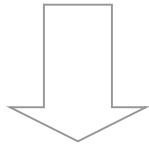
$$M = E - e \sin E = 1.3601 \text{ rad}$$

$$T = 2\pi \sqrt{\frac{a^3}{\mu}} = 2\pi \sqrt{\left( \frac{r_p + r_a}{2} \right)^3 / \mu} = 18834 \text{ s}$$

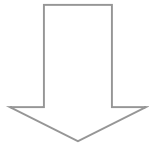
$$t = \frac{M}{2\pi} T = \frac{1.3601}{2\pi} 18834 = 4077 \text{ s} = 1\text{h}07\text{m}57\text{s}$$

# Usefulness of Kepler's Equation

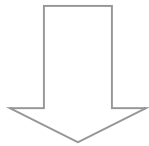
$t$  is given



$$M = \frac{2\pi t}{T}$$



$$M = E - e \sin E$$



Transcendental equation !!!  
(with a unique solution)

Practical application:

Perform a rendez-vous with the ISS (ATV, STS, Soyuz, Progress).

# Numerical Solution: Newton-Raphson

Algorithm for finding approximations to the zeros of a nonlinear function.

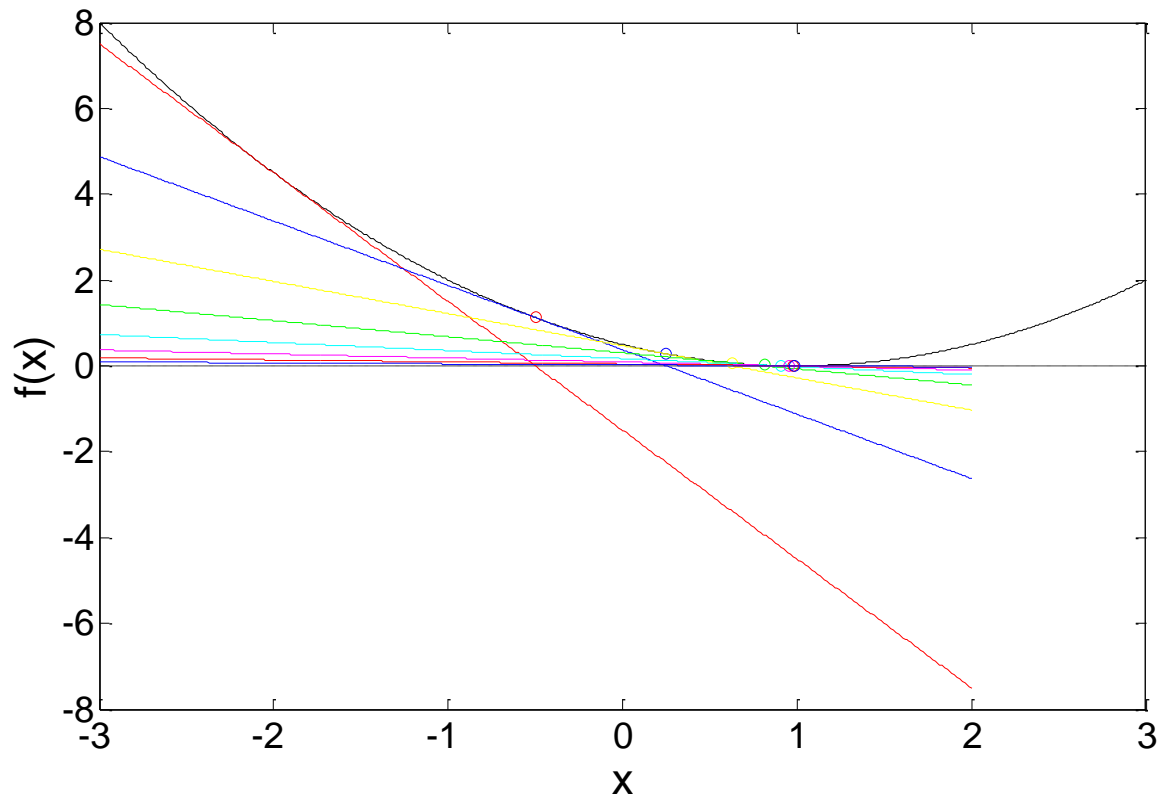
Recursive application of Taylor series truncated after the first derivative.

The initial guess should be close enough to the actual solution.



# Numerical Solution: Newton-Raphson

Example: find the zero of  $f(x)=0.5(x-1)^2$



# Analytic Solution: Lagrange

$$E = M + \sum_{n=1}^{\infty} a_n e^n$$

$$a_n = \frac{1}{2^{n-1}} \sum_{k=0}^{\text{floor}(n/2)} (-1)^k \frac{1}{(n-k)!k!} (n-2k)^{n-1} \sin[(n-2k)M]$$

Convergence if  $e < 0.663$ .

For small values of the eccentricity a good agreement with the exact solution is obtained using a few terms (e.g., 3).

# Analytic Solution: Bessel Functions

$$E = M + \sum_{n=1}^{\infty} \frac{2}{n} J_n(ne) \sin nM$$

$$J_n(x) = \sum_{k=0}^{\infty} \frac{(-1)^k}{(n+k)!k!} \left(\frac{x}{2}\right)^{n+2k}$$

Convergence for all values of the eccentricity less than 1.

# Parabolic and Hyperbolic Orbits

$$\frac{\mu^2}{h^3} t = \int_0^\theta \frac{d\theta}{(1 + \cos \theta)^2} = \frac{1}{2} \tan \frac{\theta}{2} + \frac{1}{6} \tan^3 \frac{\theta}{2}$$

$$M = \frac{\mu^2}{h^3} t = \frac{1}{2} \tan \frac{\theta}{2} + \frac{1}{6} \tan^3 \frac{\theta}{2}$$

$$F = \ln \left[ \frac{\sqrt{e+1} - \sqrt{e-1} \tan(\theta/2)}{\sqrt{e+1} + \sqrt{e-1} \tan(\theta/2)} \right]$$

$$M = e \sinh F - F$$

# Prediction of the Position and Velocity

If the position and velocity  $\mathbf{r}_0$  and  $\mathbf{v}_0$  of an orbiting body are known at a given instant  $t_0$ , how can we compute the position and velocity  $\mathbf{r}$  and  $\mathbf{v}$  at any later time  $t$ ?

Concept of  $f$  and  $g$  function and series:

$$\mathbf{r}(t) = f(t, t_0, \mathbf{r}_0, \mathbf{v}_0) \mathbf{r}_0 + g(t, t_0, \mathbf{r}_0, \mathbf{v}_0) \mathbf{v}_0$$

Prussing and Conway

$$f = 1 - \frac{a}{r_0} [1 - \cos(E - E_0)]$$

J.E. Prussing, B.A. Conway, *Orbital Mechanics*, Oxford University Press

$$g = (t - t_0) - \sqrt{\frac{a^3}{\mu}} [(E - E_0) - \sin(E - E_0)]$$

# Prediction of the Position and Velocity

Some form of Kepler's equation must still be solved by iteration. However, Gauss developed a series expansion in the elapsed time parameter  $t-t_0$ , and there is no longer the need to solve Kepler's equation:

$$\mathbf{r}(t) = f(t, t_0, \mathbf{r}_0, \mathbf{v}_0) \mathbf{r}_0 + g(t, t_0, \mathbf{r}_0, \mathbf{v}_0) \mathbf{v}_0$$

$$f = \left[ 1 - \frac{\mu}{2r_0^3} (t-t_0)^2 + \frac{\mu}{2} \frac{\mathbf{r}_0 \cdot \mathbf{v}_0}{r_0^5} (t-t_0)^3 + \dots \right]$$

$$g = \left[ (t-t_0) - \frac{\mu}{6r_0^3} (t-t_0)^3 + \frac{\mu}{4} \frac{\mathbf{r}_0 \cdot \mathbf{v}_0}{r_0^5} (t-t_0)^4 + \dots \right]$$

# The Orbit in Time



## Time systems

Clocks

Universal time

Earth's rotation

Atomic time

Coordinated universal time

Julian date

# “Everyday” Time Systems

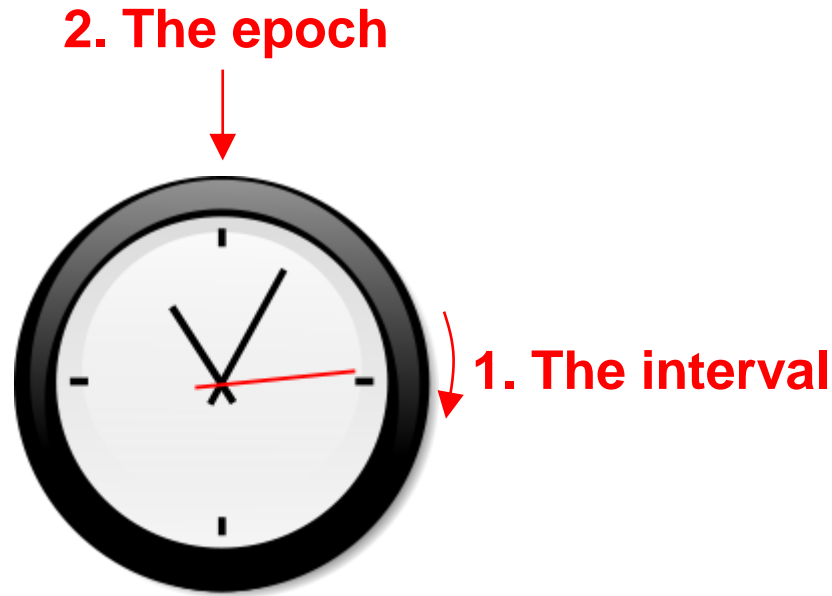
Are conventional local time systems adequate for orbital mechanics ?

- ⇒ They depend on the user’s position on Earth.
- ⇒ They are in a format (Y/M/D/H/M/S) that does not lend itself to use in a computer-implemented algorithm. For instance, what is the difference between any two dates ?

Objective of this section:  
*What would be a meaningful time system ?*



# What are the Ingredients of a Time System ?



1. The interval (a time reckoner): a repeatable phenomenon whose motion or change of state is observable and obeys a definite law.
2. The epoch (a time reference) from which to count intervals

# Historical Perspective

From remote antiquity, the celestial bodies have been the fundamental reckoners of time (e.g. rising and setting of the Sun).



Sundials were among the first instruments used to measure the time of the day. The Egyptians divided the day and night into 12h each, which varied with the seasons (unequal seasonal hour)

# Historical Perspective

---

It was not until the 14<sup>th</sup> century that an hour of uniform length became customary due to the invention of mechanical clocks.

Quartz-crystal clocks were developed in the 1920s.

The first atomic clock was constructed in 1948, and the first caesium atomic clock in 1955.

## Quantum Clocks Soon ?

# ULTRA-PRECISE QUANTUM-LOGIC CLOCK TRUMPS OLD ATOMIC CLOCK

Scientists have built a clock which is 37 times more precise than the existing international standard.

The quantum-logic clock, which detects the energy state of a single aluminum ion, keeps time to within a second every 3.7 billion years. The new timekeeper could one day improve GPS or detect the slowing of time predicted by Einstein's theory of general relativity.

<https://www.wired.com/2010/02/quantum-logic-atomic-clock/>

# Tow Important Time Scales

1. *Universal time*: the time scale based on the rotation of the Earth on its axis.
2. *Atomic time*: the time scale based on the quantum mechanics of the atom.

# Can We Use the Real Sun ?

Apparent solar time, as read directly by a sundial, is the local time defined by the actual diurnal motion of the Sun.

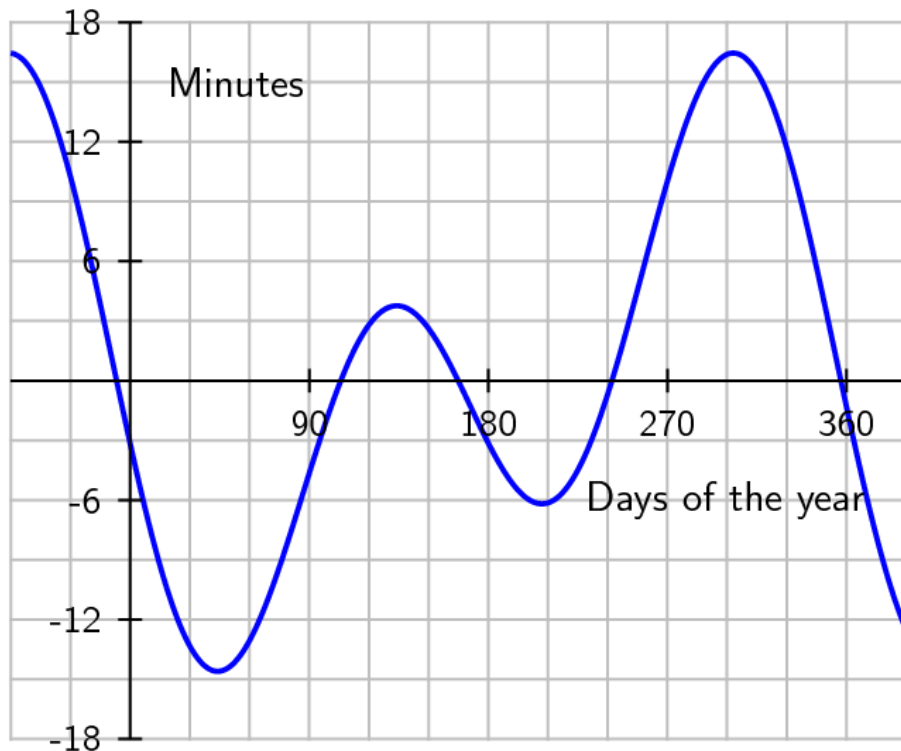
Apparent solar day is the time required for the sun to lie on the same meridian.



Due to the eccentricity of Earth's orbit, the length of the apparent solar day varies throughout the year.

⇒ The real sun is not well suited for time reckoning purposes.

# Apparent and Mean Solar Days



Approximation where  $E$  is in minutes,  $\sin$  and  $\cos$  in degrees, and  $N$  is the day number:

$$E = 9.87 \sin 2B - 7.53 \cos B - 1.5 \sin B$$

$$B = \frac{360^\circ (N - 81)}{365}$$

Equation of time

# Can We Use a Fictitious Sun ?

At noon the fictitious sun lies on the Greenwich meridian.

A mean solar day comprises 24 hours. It is the time interval between successive transits of a fictitious mean sun over a given meridian. A constant velocity in the motion about the sun is therefore assumed.

The mean solar second can be defined as  $1/86400$  of a **mean solar day**.



# Universal Time

---

Universal time is today's realization of a **mean solar time** (introduced in 1920s).

It is the same everywhere on Earth.

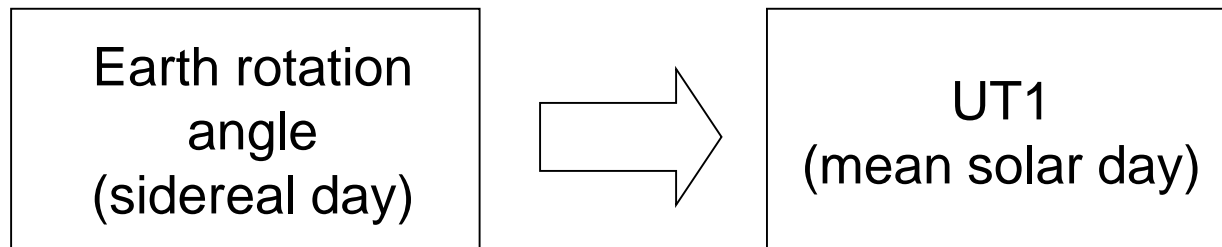
It is referred to the meridian of Greenwich and reckoned from midnight.

# Universal Time UT1

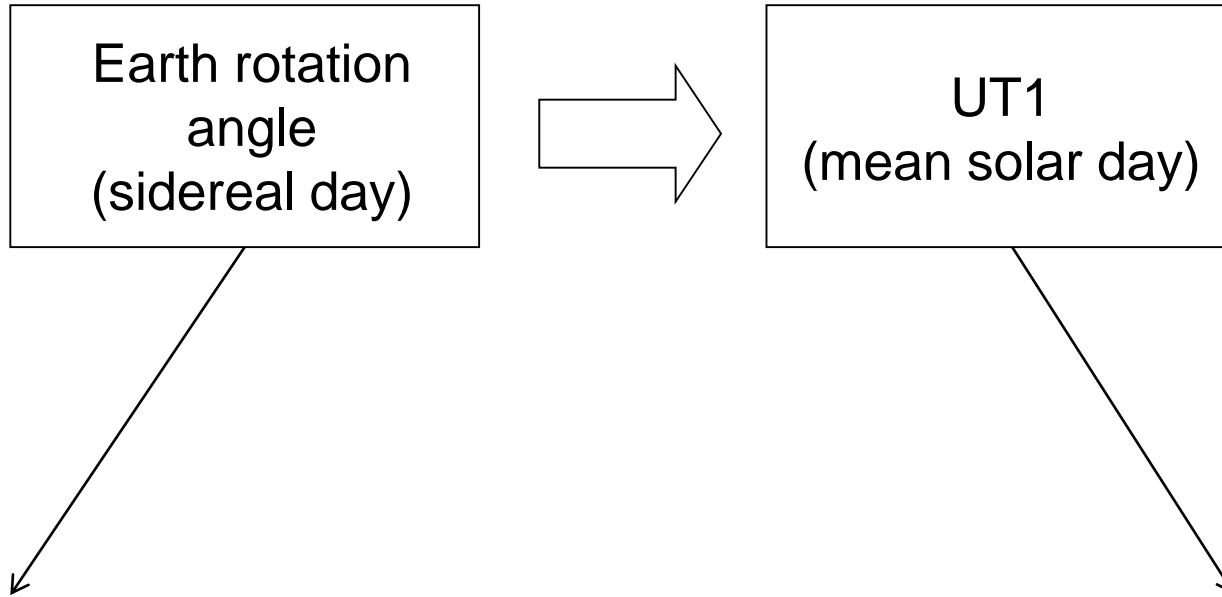
UT1 is the observed rotation of the Earth with respect to the mean sun.

It is based on the measurement of the Earth rotation angle with respect to an inertial reference frame (sidereal day).

A conversion from mean sidereal day to mean solar day is therefore necessary.



# IAU2000 Definition of Universal Time UT1



$$ERA = 2\pi(0.7790572732640 + 1.00273781191135448 Tu) \text{ radians}$$

# IAU2000 Definition of Universal Time UT1

*Earth rotation angle  
at J2000.0 UT1*

**Explanation 1**

$$ERA = 2\pi(0.7790572732640 + 1.00273781191135448 T_u) \text{ radians}$$

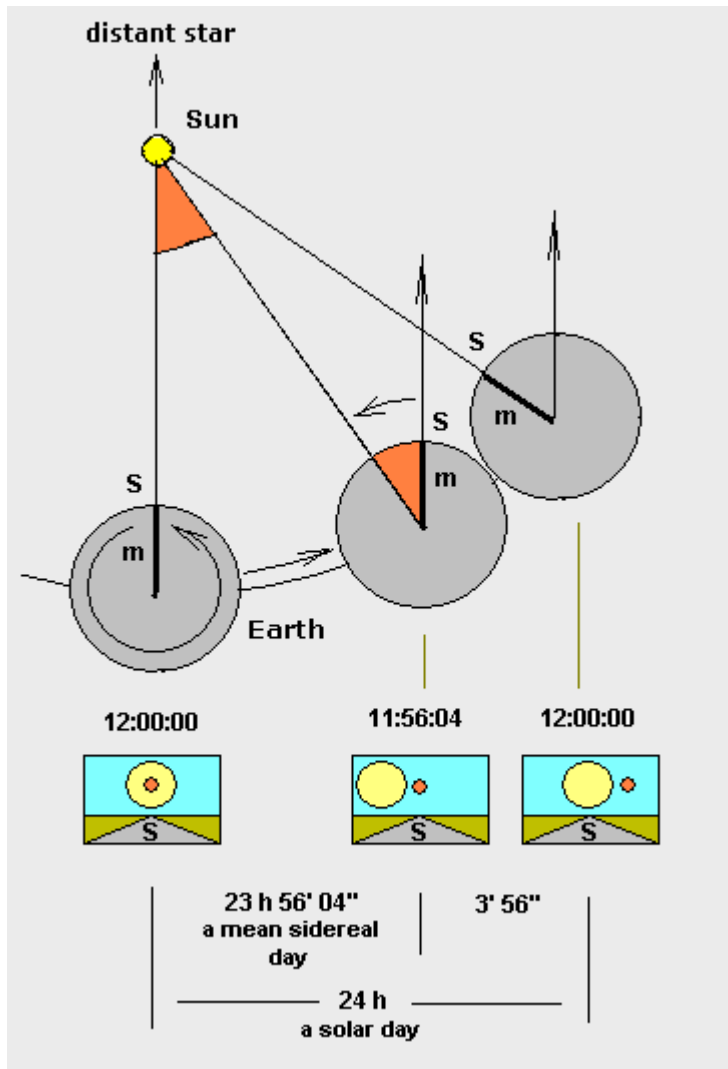
*Earth rotation angle w.r.t. ICRF.  
Its time derivative is the Earth's  
angular velocity.*

**Explanation 3**

*T<sub>u</sub> is directly related to UT1:  
Julian UT1 date - 2451545.0*

**Explanation 2**

# Explanation 1: Mean Solar Sidereal Days



1 solar day = 1.00273781191135448 sidereal day

## Explanation 2: Julian Date

The Julian day number is the number of days since noon January 1, 4713 BC → Continuous time scale and no negative dates.

For historical reasons, the Julian day begins at noon, and not midnight, so that astronomers observing the heavens at night do not have to deal with a change of date.

The number of days between two events is found by subtracting the Julian day of one from that of the other.

# Forward Computation of the Julian Date

Conversion from conventional time (YYYY, MM, DD, hh:mm:ss.ss) to Julian Date:

$$JD = 367.YYYY - \text{floor} \left\{ \frac{7}{4} \left[ YYYY + \text{floor} \left( \frac{MM + 9}{12} \right) \right] \right\} + \text{floor} \left( \frac{275.M}{9} \right) + DD + 1721013.5 + \frac{1}{24} \left[ \frac{1}{60} \left( \frac{ss}{60} + mm \right) + hh \right]$$

Valid for the period 1<sup>st</sup> March 1900 and 28<sup>th</sup> February 2100

# Backward Computation of the Julian Date

$$\begin{aligned}a &= \text{floor}\{MJD\} + 2,400,001 & MJD &= JD - 2,400,000.5 \\q &= MJD - \text{floor}\{MJD\} \\b &= \begin{cases} 0 & \text{if } a < 2299161 \\ \text{floor}\{(a - 1867216.25) / 36524.25\} & \text{otherwise} \end{cases} \\c &= \begin{cases} a + 1524 & \text{if } a < 2299161 \\ a + b - \text{floor}\{b / 4\} + 1525 & \text{otherwise} \end{cases} \\d &= \text{floor}\{(c - 121.1) / 365.25\} \\e &= \text{floor}\{365.25d\} \\f &= \text{floor}\{(c - e) / 30.6001f\} + q \\D &= c - e - \text{floor}\{30.6001f\} + q \\[D &= \text{floor}\{D\}] \\M &= f - 1 - 12 \times \text{floor}\{f / 14\} \\Y &= d - 4715 - \text{floor}\{(7 + M) / 10\} \\hr &= \text{floor}\{q \times 24\} \\min &= \text{floor}\{q \times 24 \times 60\} - (hr \times 60) \\sec &= (q \times 24 \times 60 \times 60) - (60 \times min) - (60 \times 60 \times hr)\end{aligned}$$



# Computation of Elapsed Time

Find the elapsed time between 4 October 1957 at 19:26:24 UTC and 12 May 2004 at 14:45:30 UTC

4 October 1957 at 19:26:24 UTC: 2436116.3100 days

12 May 2004 at 14:45:30 UTC: 2453138.11493056 days

→ The elapsed time is 17021.805 days

# Standard Epoch Used Today: J2000

To lessen the magnitude of the Julian date, a constant offset can be introduced. A different reference epoch 1<sup>st</sup> January 2000 at noon is used:

$$J2000 = JD - 2451545$$

# Explanation 3: Accurate Determination of ERA

The most remote objects in the universe are quasars in a distance of about 3-15 billion light years. Because quasars are at such great distances that their motions across the sky are undetectable, they form a quasi-inertial reference frame, called the international celestial reference frame.

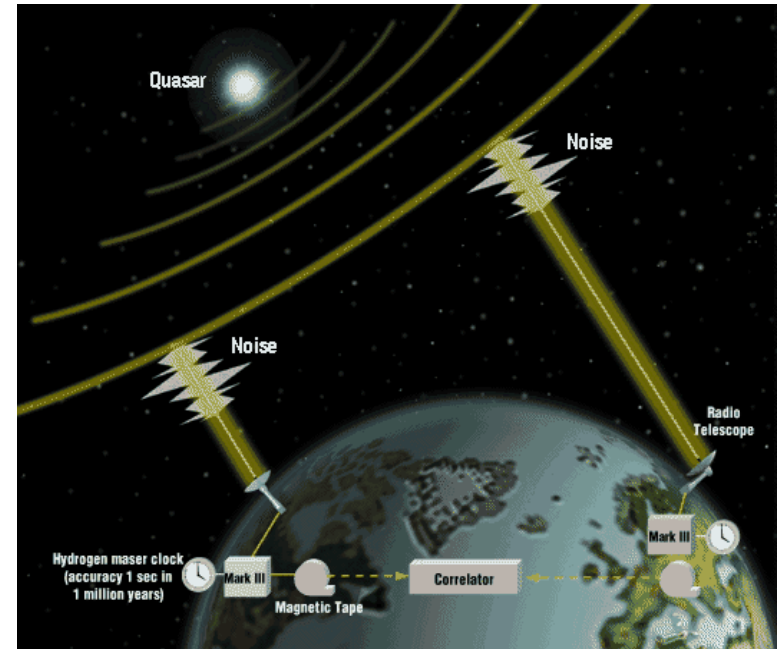
Quasars can be detected with very sensitive radiotelescopes.

By observing the diurnal motion of distant quasars (more precise than sun-based observations), it is possible to relate the position, orientation and rotation of the Earth to the inertial reference frame realized by these quasars.

# Very Long Baseline Interferometry



A radio telescope with a cryogenic dual band S/X-band receiver (TIGO, Concepcion, Chili)



# Can We Trust the Earth's Rotation ?

?

$$ERA = 2\pi(0.7790572732640 + 1.00273781191135448 Tu) \text{ radians}$$

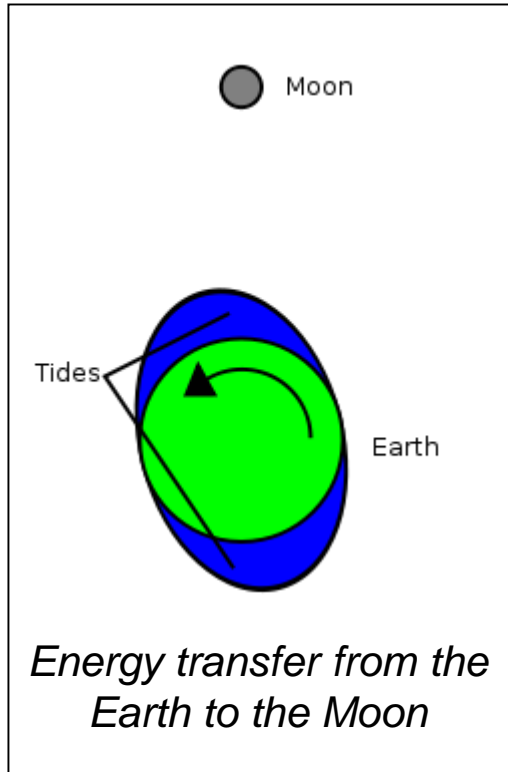
# Can We Trust the Earth's Rotation ?

No !

- ⇒ The Earth's rotation rate is not uniform. It exhibits changes on the order of 2 milliseconds per day. Corals dating from 370 millions years ago indicate that the number of days was between 385 and 410.
- ⇒ There also exists random and seasonal variations.

In addition, the axis of rotation is not fixed in space.

# Rotation Rate: Steady Deceleration (Cause 1)



The Moon is at the origin of tides: the water of the oceans bulges out along both ends of an axis passing through the centers of the Earth and Moon.

The tidal bulge closely follows the Moon in its orbit, and the Earth rotates under this bulge in a day. Due to friction, the rotation drags the position of the tidal bulge ahead of the position directly under the Moon.

A substantial amount of mass in the bulge is offset from the line through the centers of the Earth and Moon. Because of this offset, there exists a torque which boosts the Moon in its orbit, and decelerates the rotation of the Earth.

# Rotation Rate: Steady Deceleration (Cause 2)

In addition to this tidal acceleration of the Moon, the Earth is also slowing down due to tidal friction.

Tides stretch the oceans, and to a small extent, the solid mass of a planet or satellite. In one complete rotation, the planet material keeps deforming and relaxing. This takes energy away from the rotation, transforming it into heat.



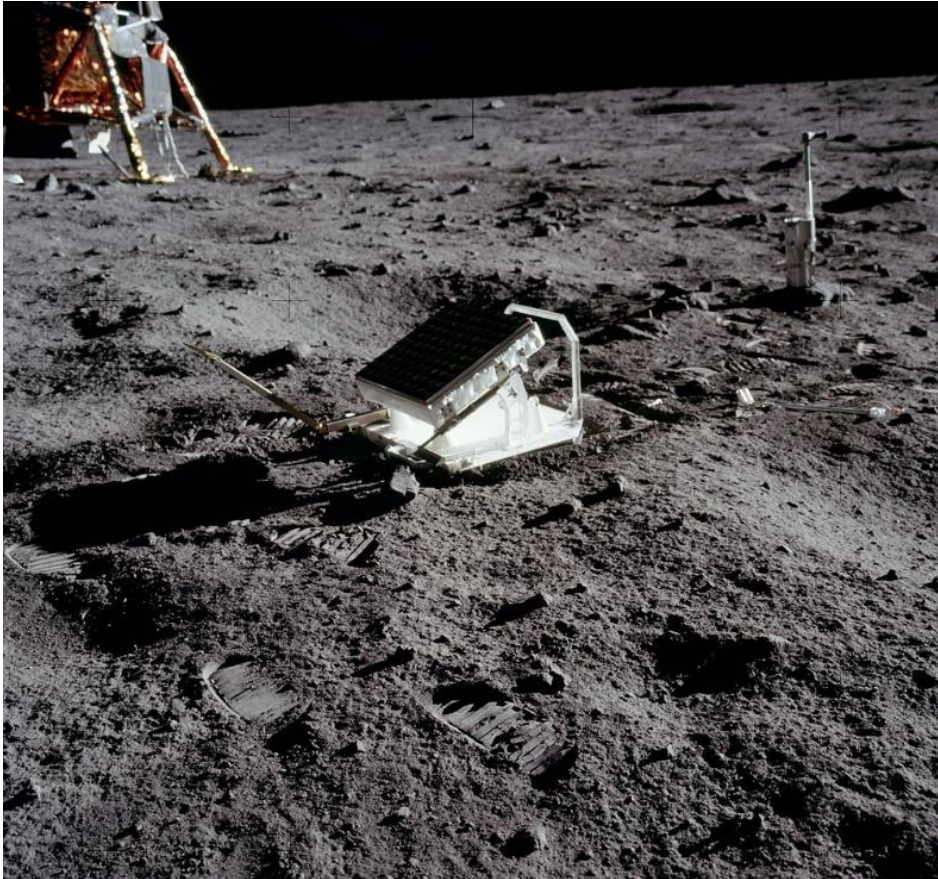
# The Moon Is Moving Away from the Earth

The secular acceleration of the Moon is small but it has a cumulative effect on the Moon's position when extrapolated over many centuries.

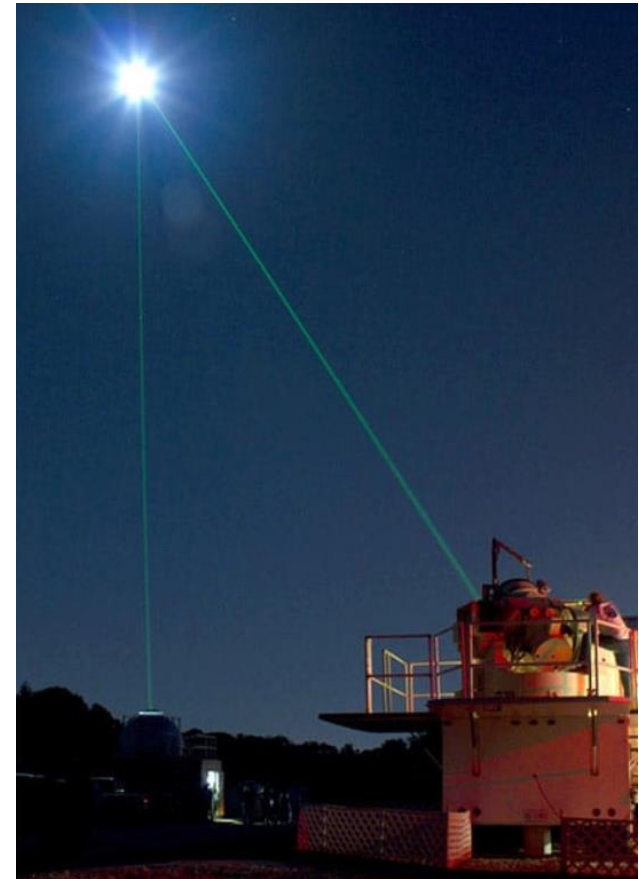
Direct measurements of the acceleration have been possible since 1969 using the Apollo retro-reflectors left on the Moon.

The results from Lunar Laser Ranging show that the Moon's mean distance from Earth is increasing by 3.8 cm per year.

# Lunar Laser Ranging Experiment (Apollo 11)



Lunar Laser Ranging Experiment from  
the Apollo 11 mission



NASA Goddard  
(Lunar Reconnaissance Orbiter)

# Le séisme au Chili pourrait avoir raccourci les jours

Rédaction en ligne

mardi 02 mars 2010, 19:48

Le violent séisme qui a frappé le Chili a peut-être fait bouger l'axe de la Terre et, par conséquent, diminué la longueur du jour terrestre. La masse de la Terre étant désormais répartie autrement, notre planète tourne plus vite.



©AFP

Le séisme de magnitude 8,8 samedi au Chili pourrait avoir eu des effets sur l'axe de rotation de la Terre, entraînant un infime raccourcissement de la durée du jour, selon des scientifiques du Jet Propulsion Laboratory (JPL) de la NASA, l'agence spatiale américaine.

D'après des calculs préliminaires, issus d'une simulation informatique, le séisme du 27 février pourrait avoir entraîné un décalage de huit centimètres de l'axe de rotation terrestre,

a expliqué mardi Richard Gross, du JPL, à Pasadena

(Californie). Cela devrait provoquer un raccourcissement des jours de 1,26 microseconde, soit 1,26 millionième de seconde, ajoute-t-il. D'après Richard Gross, l'analyse des données du séisme permettra d'affiner les calculs.

Le phénomène n'est pas inédit. Comme dans tous les séismes majeurs, la Terre peut changer sa vitesse de rotation.

Sous l'effet du séisme, la circonférence de la terre rétrécit très légèrement.

Le phénomène se retrouve lorsqu'une patineuse sur glace accélère sa rotation en fermant les bras et les rapprochant du corps. Très légère, cette accélération peut cependant être mesurée par satellite.

A titre d'exemple, le plus grand séisme du XXe siècle, d'une magnitude de 9,6 en 1960 au Chili, a fait diminuer la longueur du jour de huit microsecondes, selon une estimation des chercheurs.

Reste que l'atmosphère (frottements, jeu des masses d'air) et les marées océaniques ont une influence bien plus importante sur la durée du jour.

Quant à l'axe de la Terre, il varie naturellement tout le temps, décrivant en gros, à l'échelle d'une année, un cercle d'une dizaine de mètres. Le petit déplacement subi par l'axe de la Terre à cause du séisme chilien, estimé à huit centimètres, est donc moins élevé que le mouvement naturel de la Terre.

A ce mouvement mécanique s'ajoutent les mouvements des océans, des marées, l'influence de l'atmosphère, les éruptions volcaniques.

Pour provoquer un cataclysme et modifier réellement l'orbite terrestre, soulignent les sismologues, il faudrait une cause extérieure comme la collision avec un astéroïde.

# What is Your Conclusion ?

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We cannot “trust” the Earth’s rotation  $\Rightarrow$  the length of one second of UT1 is not constant !

Its offset from atomic time is continually changing in a not completely predictable way.

# International Atomic Time (TAI)

Since the advent of atomic time in 1955 there has been a steady transition from reliance on the Earth's rotation to the use of atomic time as the standard for the SI unit of duration (second).

The second is the duration of 9.192.631.770 cycles of the radiation corresponding to the transition between two hyperfine levels of the ground state of  $^{133}\text{Cs}$ .

Weighted average of the time kept by about 300 atomic clocks in over 50 national laboratories worldwide.

# Atomic Clocks: Stability and Accuracy



The caesium clock has high accuracy and good long-term stability.



The hydrogen maser has the best stability for periods of up to a few hours.

# Is Atomic Time the Adequate Solution ?

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**No connection with the motion of  
the sun across the sky !**



# Physical and Astronomical Times

## Astronomical clocks:

- ⇒ Related to everyday life.
- ⇒ Not consistent; the length of one second of UT is not constant. Typical accuracies  $\sim 10^{-8}$ .

## Atomic clocks:

- ⇒ Consistent. Typical accuracies  $\sim 10^{-14}$ .
- ⇒ Not related to everyday life. If no adjustment is made, then within a millennium, local noon (i.e., the local time associated with the Sun's zenith position) would occur at 13h00 and not 12h00.

# Coordinated Universal Time (UTC)

**The good practical compromise between atomic and universal times:** it is the international standard on which civil time is based.

Its **time interval** corresponds to atomic time TAI:

⇒ It is accurate.

Its **epoch** differs by no more than 0.9 sec from UT1:

⇒ The mean sun is overhead on the Greenwich meridian at noon.

# Leap Seconds

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Leap seconds were introduced in 1971 to reconcile astronomical time, which is based on the rotation of the Earth, and physical time, which can be measured with great accuracy using atomic clocks.

Leap seconds are introduced to account for the fact that the Earth currently runs slow at 2 milliseconds per day and they ensure that the Sun continues to be overhead on the Greenwich meridian at noon to within 1s.

## Le Nouvel An en retard d'une seconde cette année

Rédaction en ligne

mercredi 31 décembre 2008, 09:19

Une seconde sera ajoutée à la dernière heure de 2008 pour refléter le ralentissement de la rotation de la Terre, sur fond de débat entre partisans de deux systèmes de mesure : le Greenwich Mean Time (« heure moyenne de Greenwich », GMT), une institution britannique, et le Temps atomique international (TAI), calculé près de Paris.

Le 31 décembre, à 23 heures, 59 minutes et 59 secondes en temps universel coordonné (UTC), une seconde supplémentaire sera ajoutée.

# Not So Simple...

## Record de vitesse de rotation de la Terre : "On pourrait être amené à retirer une seconde. Cela n'est jamais arrivé"

Le mercredi 29 juin 2022, la Terre a battu son record de vitesse de rotation.



La Rédaction

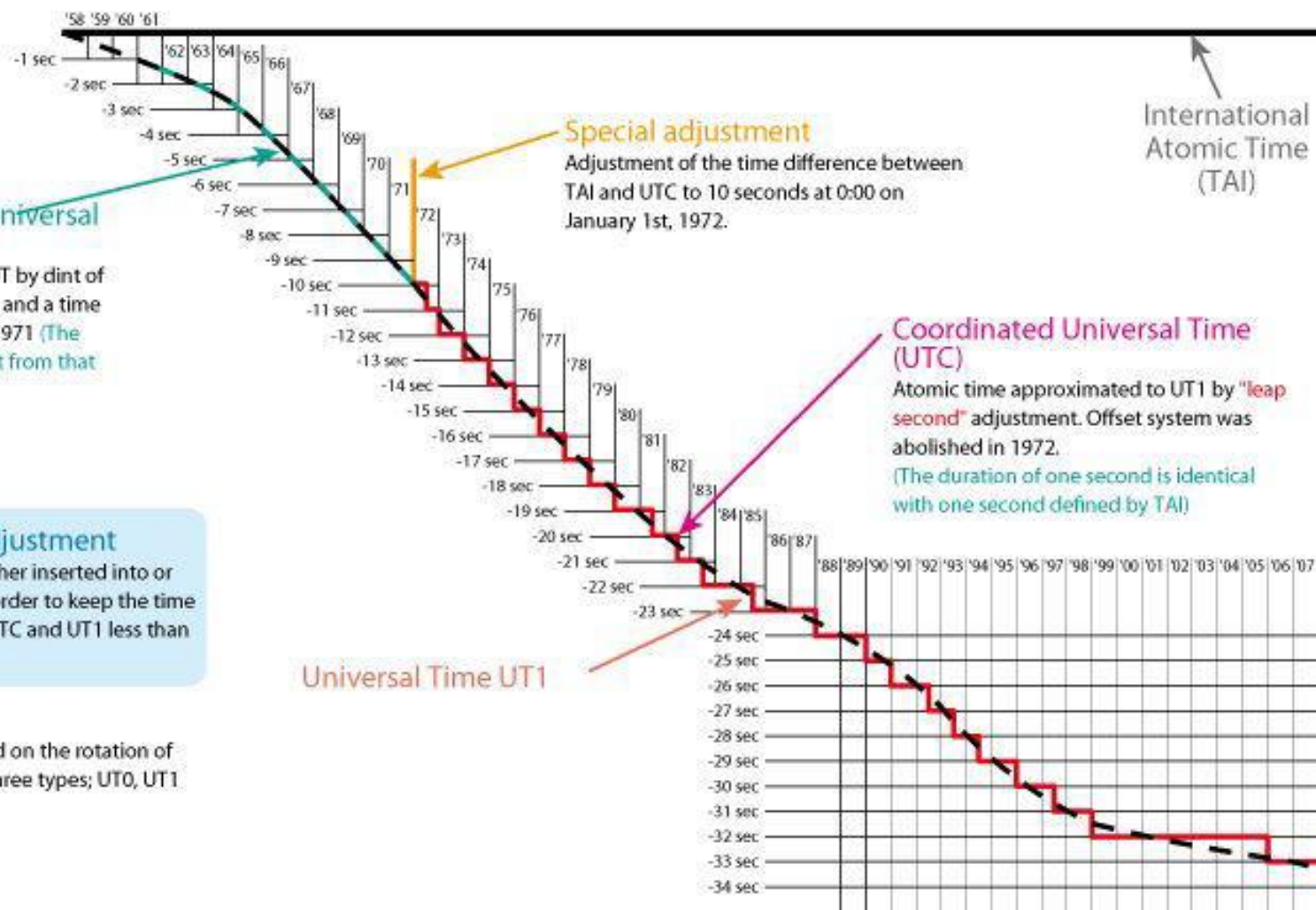


Publié le 05-08-2022 à 11h20 - Mis à jour le 05-08-2022 à 15h11



@Shutterstock

rotation de la Terre au journal Ouest-France, "depuis 2016, on s'aperçoit que la vitesse de rotation de la Terre s'accélère, et donc la durée du jour diminue". Avant cela, "depuis 1930, raconte l'astronome, on observait une baisse de la vitesse de rotation de la Terre et donc une augmentation de la durée du jour. Cette tendance s'est inversée depuis sept ans sans qu'on puisse l'expliquer".



### Former Coordinated Universal Time

Atomic time brought close to UT by dint of an offset of standard frequency and a time step adjustment from 1961 to 1971 (The duration of 1 second is different from that of TAI)

### Leap second adjustment

A step of 1 second either inserted into or deleted from UTC in order to keep the time difference between UTC and UT1 less than 0.9 seconds

### Universal Time (UT):

Time generated based on the rotation of the earth. There are three types; UT0, UT1 and UT2.

### Special adjustment

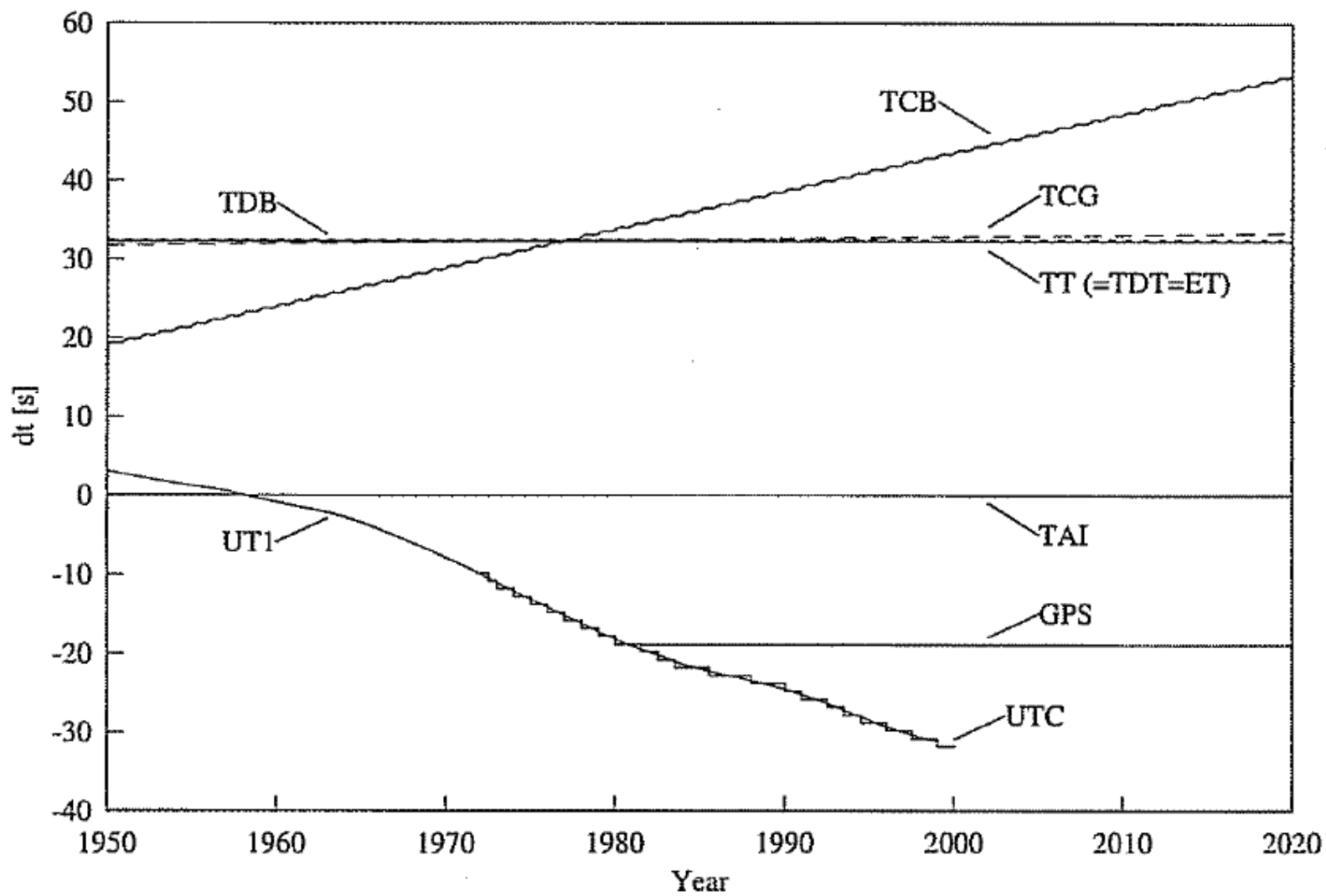
Adjustment of the time difference between TAI and UTC to 10 seconds at 0:00 on January 1st, 1972.

### Coordinated Universal Time (UTC)

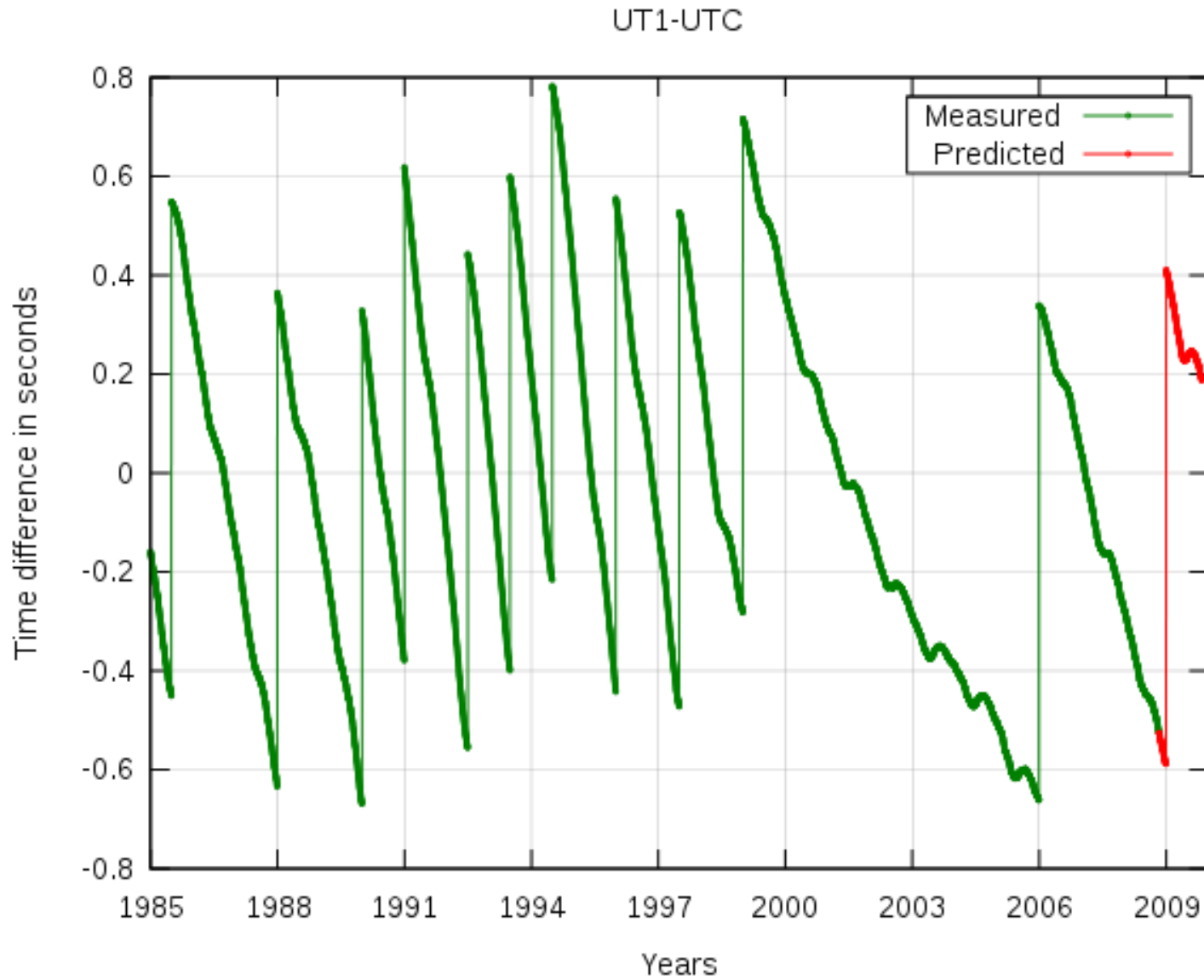
Atomic time approximated to UT1 by "leap second" adjustment. Offset system was abolished in 1972. (The duration of one second is identical with one second defined by TAI)

### Universal Time UT1

International Atomic Time (TAI)



$$|DUT1| = |UT1 - UTC| < 0.9s$$





# Leap Seconds: Pros and Cons

This is currently the subject of intense debate (UT vs. TAI; i.e., UK vs. France).

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Abandoning leap seconds would break sundials. In thousands of years, 16h00 would occur at 03h00. The British who have to wake up early in the morning to have tea...

Astronomers

-

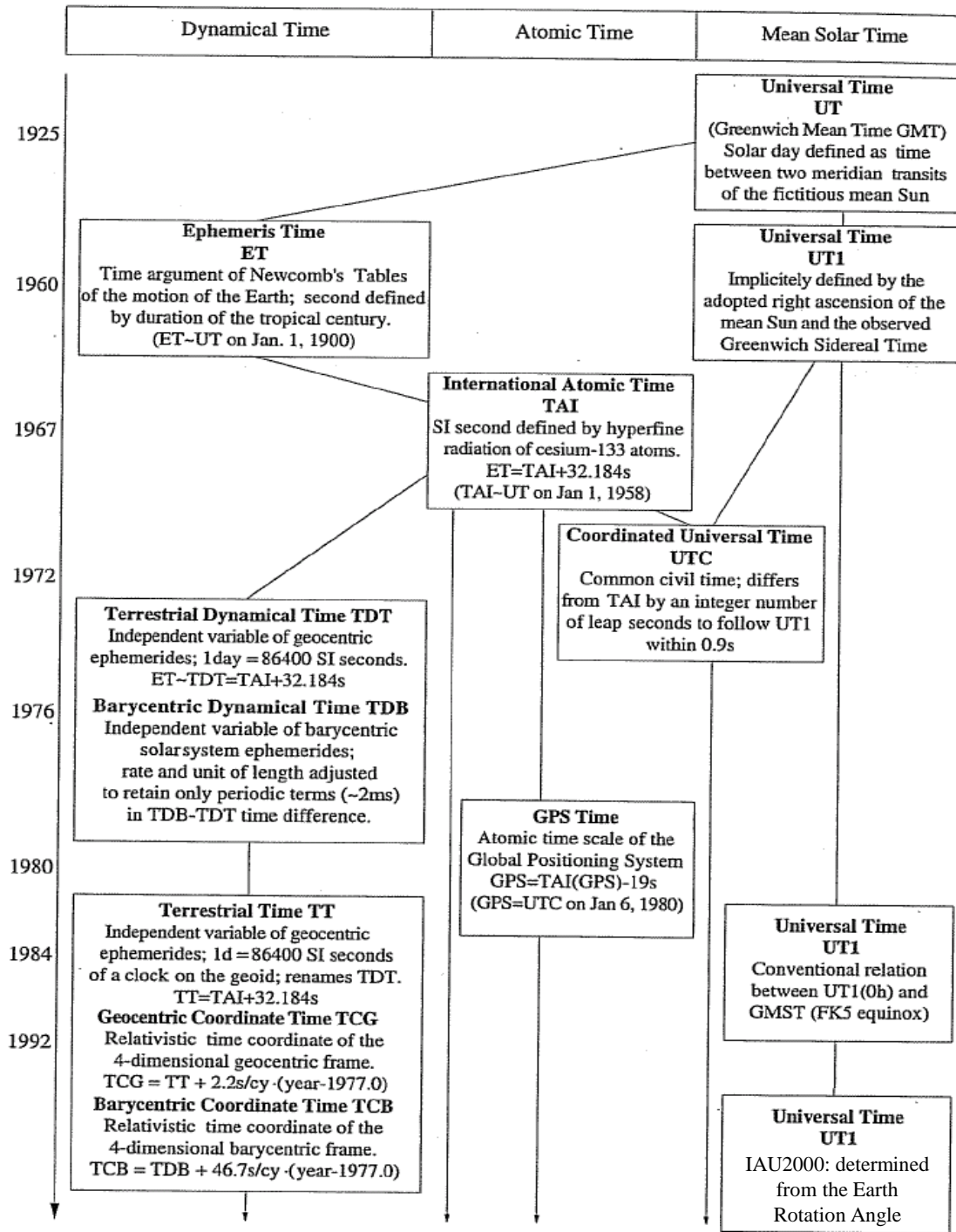
Leap seconds are a worry with safety-critical real-time systems (e.g., air-traffic control — GPS' internal atomic clocks race ahead of UTC).

# Yet More Time Systems !

GPS time: running ahead of UTC but behind TAI (it was set in 1980 based on UTC, but leap seconds were ignored since then).

Time standards for planetary motion calculations:

- ⇒ Terrestrial dynamic time: tied to TAI but with an offset of 32.184s to provide continuity with ephemeris time.
- ⇒ Barycentric dynamic time: similar to TDT but includes relativistic corrections that move the origin of the solar system barycenter.



# Further Reading

The leap second: its history and possible future

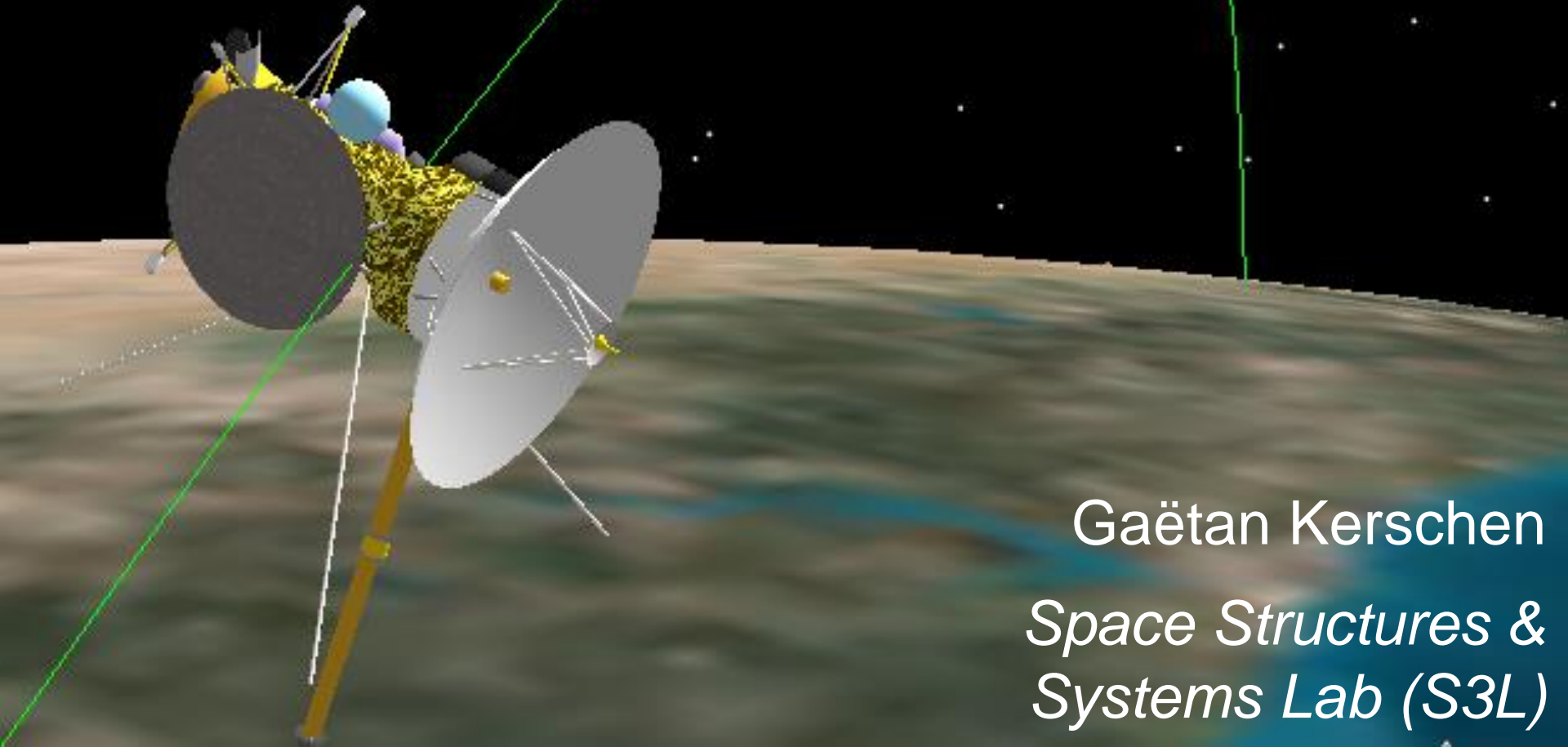
*R. A. Nelson, D. D. McCarthy, S. Malys,  
J. Levine, B. Guinot, H. F. Fliegel,  
R. L. Beard and T. R. Bartholomew*

Cassini Classical Orbit Elements  
Time (UTCG): 15 Oct 1997 09:18:54.000  
Semi-major Axis (km): 6685.637000  
Eccentricity: 0.020566  
Inclination (deg): 30.000  
RAAN (deg): 150.546  
Arg of Perigee (deg): 230.000  
True Anomaly (deg): 136.530  
Mean Anomaly (deg): 134.891

# Aerodynamics

(AERO0024)

## 3. *The Orbit in Time*



Gaëtan Kerschen  
*Space Structures &  
Systems Lab (S3L)*